Exhibit R-2, RDT&E Budget Item Justification: PB 2013 Air Force

APPROPRIATION/BUDGET ACTIVITY

3600: Research, Development, Test & Evaluation, Air Force

BA 3: Advanced Technology Development (ATD)

R-1 ITEM NOMENCLATURE

PE 0603216F: Aerospace Propulsion and Power Technology

DATE: February 2012

COST (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	FY 2014	FY 2015	FY 2016	FY 2017	Cost To Complete	Total Cost
Total Program Element	129.925	120.924	151.152	-	151.152	153.221	151.998	167.798	157.799	Continuing	Continuing
632480: Aerospace Fuels	9.091	6.768	3.581	-	3.581	2.452	4.550	4.573	4.560	Continuing	Continuing
633035: Aerospace Power Technology	5.021	5.746	3.067	-	3.067	7.520	9.892	8.944	9.601	Continuing	Continuing
634921: Aircraft Propulsion Subsystems Int	40.066	17.709	77.716	-	77.716	68.076	52.129	68.848	68.821	Continuing	Continuing
634922: Space & Missile Rocket Propulsion	29.357	27.596	22.446	-	22.446	24.061	24.388	27.598	26.631	Continuing	Continuing
635098: Advanced Aerospace Propulsion	12.744	30.117	9.553	-	9.553	18.811	42.427	39.140	29.523	Continuing	Continuing
63681B: Advanced Turbine Engine Gas Generator	33.646	32.988	34.789	-	34.789	32.301	18.612	18.695	18.663	Continuing	Continuing

A. Mission Description and Budget Item Justification

This program develops and demonstrates technologies to achieve enabling and revolutionary advances in turbine, advanced cycle, and rocket propulsion, as well as electrical power thermal management, and fuels. The program has six projects, each focusing on technologies with a high potential to enhance the performance of existing and future Air Force weapons systems. The Aerospace Fuels project develops and demonstrates improved hydrocarbon fuels and advanced propulsion systems for high-speed/hypersonic flight. The Aerospace Power Technology project develops and demonstrates power and thermal management systems for weapons and aircraft as part of energy optimized aircraft development. The Aerospace Propulsion Subsystems Integration project integrates the engine cores demonstrated in the Advanced Turbine Engine Gas Generator project with low-pressure components into demonstrator engines. The Space and Missile Rocket Propulsion project develops and demonstrates innovative rocket propulsion technologies, propellants, manufacturing techniques. The Advanced Aerospace Propulsion project develops the scramjet propulsion cycle to a technology readiness level appropriate for in-flight demonstration and for full integration with other engine cycles (including turbine and rocket based). The Advanced Turbine Engine Gas Generator project develops and demonstrates core turbine engine technologies for current and future aircraft propulsion systems. Portions of the Aerospace Fuels, Advanced Turbine Engine Gas Generator, and Aerospace Propulsion Subsystems Integration projects support adaptive cycle technology demonstrations which develop component technology for an adaptive cycle engine architecture that provides optimized performance, fuel efficiency, and durability for widely varying mission needs. Efforts in this program have been coordinated through the Reliance 21 process to harmonize efforts and eliminate duplication. This program is in Budget Activity 3, Advanced Technology Development, since it enables and demonstrates t

PE 0603216F: Aerospace Propulsion and Power Technology Air Force

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Exhibit R-2, RDT&E Budget Item Justification: PB 2013 Air Force

R-1 ITEM NOMENCLATURE

3600: Research, Development, Test & Evaluation, Air Force

PE 0603216F: Aerospace Propulsion and Power Technology

DATE: February 2012

BA 3: Advanced Technology Development (ATD)

APPROPRIATION/BUDGET ACTIVITY

B. Program Change Summary (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
Previous President's Budget	136.135	120.953	113.398	_	113.398
Current President's Budget	129.925	120.924	151.152	-	151.152
Total Adjustments	-6.210	-0.029	37.754	-	37.754
 Congressional General Reductions 	-	-0.029			
 Congressional Directed Reductions 	-	-			
 Congressional Rescissions 	-	-			
Congressional Adds	-	-			
 Congressional Directed Transfers 	-	-			
Reprogrammings	-	-			
SBIR/STTR Transfer	-3.494	-			
Other Adjustments	-2.716	-	37.754	-	37.754

Change Summary Explanation

FY11: Other Adjustments include -0.716 Congressional General Reductions and -2.000 Congressional Directed Transfers

FY13: Increase due to higher Air Force priorities for adaptive turbine engine technologies

PE 0603216F: Aerospace Propulsion and Power Technology Air Force

Exhibit R-2A, RDT&E Project Jus	tification: PE	3 2013 Air Fo	orce						DATE: Febr	uary 2012	
APPROPRIATION/BUDGET ACTIV	/ITY			R-1 ITEM N	IOMENCLAT	ΓURE		PROJECT	FY 2017 Comple		
3600: Research, Development, Tes	t & Evaluatio	n, Air Force		PE 060321	6F: <i>Aerospa</i>	ce Propulsio	n and	632480: Ae	rospace Fue	ls	
BA 3: Advanced Technology Develo	opment (ATD))		Power Tech	nology						
COST (f in Milliana)			FY 2013	FY 2013	FY 2013					Cost To	
COST (\$ in Millions)	FY 2011	FY 2012	Base	oco	Total	FY 2014	FY 2015	FY 2016	FY 2017	Complete	Total Cost
632480: Aerospace Fuels	9.091	6.768	3.581	-	3.581	2.452	4.550	4.573	4.560	Continuing	Continuing

A. Mission Description and Budget Item Justification

This project evaluates and demonstrates improved hydrocarbon fuels, unique/alternate fuels and advanced, novel aerospace propulsion technologies for Air Force applications; including high-speed/hypersonic flight and technologies to increase turbine engine operational reliability, durability, mission flexibility, and performance while reducing weight, fuel consumption, and cost of ownership. The advanced fuel emphasis is on demonstrating new thermally stable, high-heat sink, and controlled chemically reacting fuels for a conventional turbine engine, turbine-based combined cycle engines, and other advanced propulsion systems. The project also evaluates and demonstrates fuel system components that minimize cost, reduce maintenance, and improve performance of future aerospace systems. The advanced propulsion emphasis is on demonstrating concepts for combined cycle, ramjet, and scramjet engines. A portion of this project supports the demonstration of adaptive cycle technologies. This project develops component technology for an adaptive cycle engine architecture that provides optimized performance, fuel efficiency, and durability for widely varying mission needs.

B. Accomplishments/Planned Programs (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
Title: Major Thrust 1.	2.866	1.000	-	-	-
Description: Demonstrate thermally stable fuels and fuel system hardware concepts to enhance cooling capacity (performance), minimize fuel coking, and reduce fuel system maintenance.					
FY 2011 Accomplishments: Demonstrated adaptive engine cycles for high efficiency and ultra efficient turbine engine technologies integrated power/thermal management systems that included cooled cooling air systems, as well as approaches to					
deoxygenate fuel to improve thermal stability.					
FY 2012 Plans: Demonstrate advanced fuel-based turbine engine cooling approaches. Note: In FY 2012, efforts in this thrust are decreased due to higher Air Force (AF) priorities.					
FY 2013 Base Plans: Decrease in FY 2013 due to higher Department of Defense priorities.					
FY 2013 OCO Plans: N/A					
Title: Major Thrust 2.	1.196	1.000	-	-	-

PE 0603216F: Aerospace Propulsion and Power Technology Air Force

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			D	ATE: Febru	ary 2012	
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 3: Advanced Technology Development (ATD)	R-1 ITEM NOMENCLATURE PE 0603216F: Aerospace Propulsion ar Power Technology		PROJECT 332480: Aeros	rospace Fuels		
B. Accomplishments/Planned Programs (\$ in Millions)		FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
Description: Develop and demonstrate efficacy of low-cost, environand reduce soot/particulate emissions from gas turbine engines.	onmentally friendly fuel approaches to assess					
FY 2011 Accomplishments: Assessed fuel structure/combustion performance relationship in hig effectiveness of chemical kinetic models for jet fuels to match high						
FY 2012 Plans: Demonstrate state-of-the-art soot/particulate diagnostics in full scale	le engine testing.					
FY 2013 Base Plans: Decrease in FY 2013 due to higher Department of Defense priorities	es.					
FY 2013 OCO Plans: N/A						
Title: Major Thrust 3.		1.04	3 1.000	-	-	-
Description: Develop and demonstrate enhancements to fuel syst	em technology.					
FY 2011 Accomplishments: Developed advanced fuel catalyst and composition approaches to sink goals.	achieve 2nd generation endothermic fuel heat					
FY 2012 Plans: Demonstrate effective structural cooling of 2nd generation endother	ermic fuels for hypersonic vehicles.					
FY 2013 Base Plans: Decrease in FY 2013 due to higher Department of Defense priorities	es.					
FY 2013 OCO Plans: N/A						
Title: Major Thrust 4.		1.09	0.770	3.581	-	3.581
Description: Identify, develop, and demonstrate low-cost approach the Expeditionary Air Force.	hes to reducing the fuel logistics footprint for					
FY 2011 Accomplishments:						

PE 0603216F: Aerospace Propulsion and Power Technology Air Force UNCLASSIFIED

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Exhibit R-2A, RDT&E Project Jus	stification: PB	2013 Air Fo	rce					D	ATE: Febr	uary 2012		
APPROPRIATION/BUDGET ACTI 3600: Research, Development, Tes BA 3: Advanced Technology Devel	st & Evaluation,	Air Force		R-1 ITEM NC PE 0603216I Power Techn	: E: Aerospac	URE e Propulsion a		FY 2013 FY 2013 F				
B. Accomplishments/Planned Pr	ograms (\$ in N	Millions)					FY 2011	FY 2012			FY 2013 Total	
Modeled spread of biological mate advanced additives for mitigation of) through fue	el handling sy	rstems. Den	nonstrated						
FY 2012 Plans: Evaluate fuel compositional relation	nship to biologi	cal growth.										
FY 2013 Base Plans: Demonstrate mitigation of biological distribution systems.	al growth in alte	ernative fuel	s and comm	ercial jet fuel	s in base-le	vel fuel						
FY 2013 OCO Plans: N/A												
Title: Major Thrust 5.							2.88	9 2.998	-	-	-	
Description: Characterize and der certifications and standards for jet		ise of altern	ative hydroca	arbon jet fue	I to comply v	with Air Force						
FY 2011 Accomplishments: Demonstrated biomass-derived fue feedstocks. Studied greenhouse ga						m varying						
FY 2012 Plans: Demonstrate improved alternative Evaluate fully-synthetic biofuels in						cation.						
FY 2013 Base Plans: Decrease in FY 2013 due to higher	r Department o	f Defense p	riorities.									
FY 2013 OCO Plans: N/A												
			Accomplis	hments/Plar	ned Progra	ams Subtotal	s 9.09	1 6.768	3.581	-	3.581	
C. Other Program Funding Sumn	mary (\$ in Milli	ons)	EV 2042	EV 2042	EV 2042					Cost To		
l in a Mann	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	FY 2014	FY 2015	FY 2016	FY 2017	Complete	Total Cost	
Line Item	=v . 1										iotai occi	

PE 0603216F: Aerospace Propulsion and Power Technology Air Force UNCLASSIFIED Page 5 of 23

Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			DATE: February 2012
APPROPRIATION/BUDGET ACTIVITY	R-1 ITEM NOMENCLATURE	PROJECT	
3600: Research, Development, Test & Evaluation, Air Force	PE 0603216F: Aerospace Propulsion and	632480: Aei	rospace Fuels
BA 3: Advanced Technology Development (ATD)	Power Technology		

D. Acquisition Strategy

N/A

E. Performance Metrics

Please refer to the Performance Base Budget Overview Book for information on how Air Force resources are applied and how those resources are contributing to Air Force performance goals and most importantly, how they contribute to our mission.

PE 0603216F: Aerospace Propulsion and Power Technology Air Force

Exhibit R-2A, RDT&E Project Just	tification: PE	3 2013 Air Fo	orce						DATE: Febr	uary 2012		
APPROPRIATION/BUDGET ACTIV 3600: Research, Development, Test BA 3: Advanced Technology Develo	t & Evaluation			R-1 ITEM N PE 0603216 Power Tech	6F: Aerospa		n and	PROJECT 633035: <i>Ae</i>	rospace Pov	ospace Power Technology		
COST (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	FY 2014	FY 2015	FY 2016	FY 2017	Cost To Complete	Total Cost	
633035: Aerospace Power Technology	5.021	5.746	3.067	-	3.067	7.520	9.892	8.944	9.601	Continuing	Continuing	

A. Mission Description and Budget Item Justification

This project develops and demonstrates electrical power, thermal management, and distribution for aerospace applications. This technology enhances reliability and survivability, and reduces vulnerability, weight, and life cycle costs for manned and remotely piloted aircraft. The electrical power system components developed are projected to provide a two- to five-fold improvement in aircraft reliability and maintainability, and a reduction in power system weight. This project is integrated into energy optimized aircraft efforts and power and thermal programs. This project also develops and demonstrates electrical power and thermal management technologies to enable solid state high power density sources for directed energy weapons.

B. Accomplishments/Planned Programs (\$ in Millions)			FY 2013	FY 2013	FY 2013
	FY 2011	FY 2012	Base	oco	Total
Title: Major Thrust 1.	0.250	0.200	-	-	-
Description: Develop electrical power and thermal management component subsystem technologies for integration with directed energy weapons (DEW) to deliver high power for DEW operation.					
FY 2011 Accomplishments: Supported development of energy storage, power conditioning, and thermal management subsystems to support flight demonstration of a high energy laser.					
FY 2012 Plans: Support integration of power and thermal management subsystems for flight demonstration of a high energy laser.					
FY 2013 Base Plans: Decrease in FY 2013 due to higher Department of Defense priorities.					
FY 2013 OCO Plans: N/A					
Title: Major Thrust 2.	4.287	5.055	3.067	-	3.067
Description: Develop power generation/conditioning/distribution component, energy storage, and thermal management components and subsystem technologies for integration into high power aircraft.					
FY 2011 Accomplishments:					

PE 0603216F: Aerospace Propulsion and Power Technology Air Force

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			D	ATE: Febru	ary 2012		
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 3: Advanced Technology Development (ATD)	R-1 ITEM NOMENCLATURE PE 0603216F: Aerospace Propulsion an Power Technology	PROJECT 633035: Aerospace Power Technology					
B. Accomplishments/Planned Programs (\$ in Millions)		FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	
Integrated, fabricated, and modified high temperature, energy optin components. Integrated subsystems (including rugged/robust power performance electric actuators, and adaptive power and thermal mintegrated system level evaluation testing. Performed system modified integrated subsystems meet design criteria and performance objectives.	er electronics, motor controls, high nanagement technologies) and performed ifications as necessary to demonstrate that						
FY 2012 Plans: Demonstrate robust, high power, high temperature power and ther hardware in the loop validation and verification of system level energy.							
FY 2013 Base Plans: Demonstrate adaptive power and thermal management subsystem system level energy optimized air platform models.	ns for next generation air platforms based on						
FY 2013 OCO Plans: N/A							
Title: Major Thrust 3.		0.48	4 0.491	-	-	-	
Description: Develop hybrid electrical power and thermal manage technologies for special purpose applications, enabling long endur FY 2011 Accomplishments: Developed and fabricated energy optimized, lightweight, hybrid electrical power and thermal manage technologies for special purpose applications, enabling long endur	ance small remotely piloted aircraft (RPA).						
subsystems for increased endurance RPA and ground based spec							
FY 2012 Plans: Demonstrate ruggedized high endurance small RPA hybrid power	and thermal management subsystems.						
FY 2013 Base Plans: Decrease in FY 2013 due to higher Department of Defense prioritie	es.						
FY 2013 OCO Plans: N/A							
Acc	omplishments/Planned Programs Subtotals	5.02	1 5.746	3.067	-	3.067	

PE 0603216F: Aerospace Propulsion and Power Technology Air Force UNCLASSIFIED Page 8 of 23

Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force

APPROPRIATION/BUDGET ACTIVITY

3600: Research, Development, Test & Evaluation, Air Force
BA 3: Advanced Technology Development (ATD)

BA 5: Advanced Technology Development (ATD)

DATE: February 2012

R-1 ITEM NOMENCLATURE
PE 0603216F: Aerospace Propulsion and Power Technology

C. Other Program Funding Summary (\$ in Millions)

			FY 2013	FY 2013	FY 2013					Cost To	
<u>Line Item</u>	FY 2011	FY 2012	Base	OCO	<u>Total</u>	FY 2014	FY 2015	FY 2016	FY 2017	Complete	Total Cost
• N/A: <i>N/A</i>	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	Continuing	Continuing

D. Acquisition Strategy

N/A

E. Performance Metrics

Please refer to the Performance Base Budget Overview Book for information on how Air Force resources are applied and how those resources are contributing to Air Force performance goals and most importantly, how they contribute to our mission.

PE 0603216F: Aerospace Propulsion and Power Technology Air Force

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force										DATE: February 2012			
APPROPRIATION/BUDGET ACTIV 3600: Research, Development, Test BA 3: Advanced Technology Develo	t & Evaluation			R-1 ITEM N PE 0603210 Power Tech	6F: <i>Aerospa</i>		n and	PROJECT 634921: <i>Air</i>	PROJECT 634921: Aircraft Propulsion Subsystems Int				
COST (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	FY 2014	FY 2015	FY 2016	FY 2017	Cost To Complete	Total Cost		
634921: Aircraft Propulsion Subsystems Int	40.066	17.709	77.716	-	77.716	68.076	52.129	68.848	68.821	Continuing	Continuing		

A. Mission Description and Budget Item Justification

This project develops and demonstrates technology to increase turbine engine operational reliability, durability, mission flexibility, and performance while reducing weight, fuel consumption, and cost of ownership. The Aerospace Propulsion Subsystems Integration (APSI) project includes demonstrator engines for manned systems and concept and efficient small-scale propulsion for remotely piloted aircraft and cruise missile applications. The demonstrator engines integrate the core (high-pressure spool) technology developed under the Advanced Turbine Engine Gas Generator project with the engine (low-pressure spool) technology such as fans, turbines, engine controls, mechanical systems, exhaust nozzles, and augmentors. Additionally, this project includes activities to improve propulsion safety and readiness. This project also focuses on integration of inlets, nozzles, engine/airframe compatibility, and power and thermal management subsystems technologies. The APSI project provides aircraft with potential for longer range and higher cruise speeds with lower specific fuel consumption, surge power for successful engagements, high sortie rates with reduced maintenance, reduced life cycle cost, and improved survivability, resulting in increased mission effectiveness. Technologies developed are applicable to sustained high-speed vehicles and responsive space launch. The APSI project is focused on improving propulsion capabilities while at the same time reducing the cost of ownership. Anticipated technology advances include turbine engine improvements providing approximately twice the range for a sustained supersonic combat aircraft, doubling the time on station with 10 times the power output for surveillance aircraft and propulsion for a high speed supersonic missile with double the range for time sensitive targets. A portion of this project supports the demonstration of adaptive cycle technologies, which develop component technology for an adaptive cycle engine architecture that provides optimized performance, fuel effic

B. Accomplishments/Planned Programs (\$ in Millions)			FY 2013	FY 2013	FY 2013
	FY 2011	FY 2012	Base	oco	Total
Title: Major Thrust 1.	7.267	1.800	0.500	-	0.500
Description: Design, fabricate, and demonstrate durability and integration technologies for turbofan/turbojet engines to improve durability, supportability, and affordability of AF aircraft.					
FY 2011 Accomplishments: Completed detailed design and began fabricate hardware for advanced features for durable fans, turbines, mechanical systems, interactions between the inlet and fan, and controls/accessories, to include advanced cooling design for low pressure turbine blades, health monitoring, light weight externals, and repair validation.					
FY 2012 Plans:					

PE 0603216F: Aerospace Propulsion and Power Technology Air Force

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			D	ATE: Febru	ary 2012	
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 3: Advanced Technology Development (ATD)	nd 63	on Subsyste	ems Int			
B. Accomplishments/Planned Programs (\$ in Millions)		FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
Continue fabrication of low spool engine components. Investigate inlet a to develop controls and accessories, health monitoring technologies, are Continue to assess and validate repair techniques.						
FY 2013 Base Plans: Investigate inlet and exhaust interactions. Decrease in FY 2013 due to	higher Department of Defense priorities.					
FY 2013 OCO Plans: N/A						
Title: Major Thrust 2.		24.805	14.709	9.167	-	9.167
Description: Design, fabricate, and test advanced component technoloconsumption of turbofan/turbojet engines.	ogies for improved performance and fuel					
FY 2011 Accomplishments: Continued fabrication and began assembly of advanced adaptive cycle including an advanced fan, high work variable low turbine for long dwell advanced exhaust nozzle for subsonic to sustained supersonic flight. Cobypass/high overall pressure ratio engine technologies for improved fue	time, controls, inlet integration, and ontinued preliminary design for a high					
FY 2012 Plans: Complete assembly and instrumentation of advanced adaptive cycle (the including an advanced fan, high work variable low turbine for long dwell advanced exhaust nozzle for subsonic to sustained supersonic flight. Contechnologies. Continue preliminary design of high bypass/high overall primproved fuel consumption. Note: In FY 2012, efforts in this thrust are design of high bypass/high overall primproved fuel consumption.	time, controls, inlet integration, and Conduct experimental testing of engine ressure ratio engine technologies for					
FY 2013 Base Plans: Finish assembly and instrumentation of advanced adaptive cycle (third an advanced fan, high work variable low turbine for long dwell time, con exhaust nozzle for subsonic to sustained supersonic flight. Decrease in Defense priorities.	ntrols, inlet integration, and advanced					
FY 2013 OCO Plans: N/A						
Title: Major Thrust 3.		7.994	1.200	15.916	-	15.916

PE 0603216F: Aerospace Propulsion and Power Technology Air Force UNCLASSIFIED
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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			D	ATE: Febru	ary 2012		
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 3: Advanced Technology Development (ATD)	PROJECT 634921: Aircraft Propulsion Subsystems Int						
B. Accomplishments/Planned Programs (\$ in Millions)		FY 201	1 FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	
Description: Design, fabricate, and test component technologies for performance, durability, and affordability of missile and remotely pilo	• •						
FY 2011 Accomplishments: Conducted detailed design of a higher specific thrust, low-cost experience efficiency to significantly improve range. Conducted detailed design spool, and advanced engine components for fuel efficient subsonic to the components of the comp	of advanced fan, advanced low spool turbine						
FY 2012 Plans: Complete detailed design and initiate fabrication of components of a turbine engine for improved fuel efficiency to significantly improve ra fan, advanced low spool turbine spool, and advanced engine compo	nge. Continue detailed design of advanced nents for fuel efficient subsonic unmanned						
FY 2013 Base Plans: Complete assembly and instrumentation of supersonic, long endural technology rig testing and begin sea level testing of supersonic, long engine activity to meet follow on activity need date).							
Title: Major Thrust 4.				52.133	-	52.133	
Description: Design, fabricate, and demonstrate performance, dura adaptive turbine engine technologies.	bility, and operability technologies to mature						
FY 2011 Accomplishments: N/A							
FY 2012 Plans: N/A							
FY 2013 Base Plans: Complete preliminary designs for an adaptive turbine engine with recthrust-to-weight, and reduced cost. Perform augmentor/exhaust noz design of an advanced adaptive fan.							
Accor	mplishments/Planned Programs Subtotals	40.06	17.709	77.716	-	77.716	

PE 0603216F: Aerospace Propulsion and Power Technology Air Force UNCLASSIFIED
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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			DATE: February 2012
APPROPRIATION/BUDGET ACTIVITY	R-1 ITEM NOMENCLATURE	PROJECT	
3600: Research, Development, Test & Evaluation, Air Force	PE 0603216F: Aerospace Propulsion and	634921: Aii	rcraft Propulsion Subsystems Int
BA 3: Advanced Technology Development (ATD)	Power Technology		

C. Other Program Funding Summary (\$ in Millions)

			FY 2013	FY 2013	FY 2013					Cost To	
<u>Line Item</u>	FY 2011	FY 2012	Base	OCO	<u>Total</u>	FY 2014	FY 2015	FY 2016	FY 2017	Complete	Total Cost
• N/A: <i>N/A</i>	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	Continuing	Continuing

D. Acquisition Strategy

N/A

E. Performance Metrics

Please refer to the Performance Base Budget Overview Book for information on how Air Force resources are applied and how those resources are contributing to Air Force performance goals and most importantly, how they contribute to our mission.

PE 0603216F: Aerospace Propulsion and Power Technology Air Force

Exhibit R-2A, RDT&E Project Just	orce						DATE: Feb	uary 2012			
APPROPRIATION/BUDGET ACTIV 3600: Research, Development, Test BA 3: Advanced Technology Develo		R-1 ITEM N PE 0603216 Power Tech	6F: Aerospa		n and	PROJECT 634922: <i>Sp</i>	opulsion				
COST (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	FY 2014	FY 2015	FY 2016	FY 2017	Cost To Complete	Total Cost
634922: Space & Missile Rocket Propulsion	29.357	27.596	22.446	-	22.446	24.061	24.388	27.598	26.631	Continuing	Continuing

A. Mission Description and Budget Item Justification

This project develops and demonstrates advanced and innovative low-cost rocket turbo-machinery and components, low-cost space launch propulsion technologies, and advanced propellants for launch and orbit transfer propulsion. Additionally, this project develops technologies for the sustainment of strategic systems (including solid boost/missile propulsion, post boost control, and aging and surveillance efforts) and tactical rockets. Characteristics such as environmental acceptability, affordability, reliability, responsiveness, reduced weight, and reduced operation and launch costs are emphasized. Increased life and performance of propulsion systems are key goals. This project also develops chemical, electrical, and solar rocket propulsion technologies for station-keeping and on-orbit maneuvering applications. Technology areas investigated include ground demonstrations of compact, lightweight, advanced propulsion technologies, higher efficiency energy conversion systems (derived from an improved understanding of combustion fundamentals), and high-energy propellants. Technological advances developed in this program could improve the performance of expendable payload capabilities by approximately 20-50 percent and reduce launch, operations, and support costs by approximately 30 percent. Responsiveness and operability of propulsion systems will be enhanced for reusable launch systems. Technology advances could also lead to seven-year increase in satellite on-orbit time, a 50 percent increase in satellite maneuvering capability, a 25 percent reduction in orbit transfer operational costs, and a 15 percent increase in satellite payload. Aging and surveillance efforts for solid rocket motors could reduce lifetime prediction uncertainties for individual motors by 50 percent, enabling motor replacement for cause. The efforts in this project contribute to the sustainment of the rocket propulsion industry, providing rocket propulsion technology for the entire Department of Defense and often NASA.

B. Accomplishments/Planned Programs (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
Title: Major Thrust 1.	22.210	20.860	19.680	-	19.680
Description: Develop liquid rocket propulsion technology for current and future space launch vehicles.					
FY 2011 Accomplishments: Continued, through hot fire testing, the validation and verification of modeling and simulation tools developed for advanced cryogenic upper stage technologies. Continued development of hydrocarbon engine components for integration and demonstration in an advanced hydrocarbon engine concept for future reusable launch vehicles. Continued sub-scale component testing to demonstrate hydrocarbon boost technologies. Continued material manufacturing scale-up effort to support hydrocarbon boost demonstration program.					
FY 2012 Plans: Complete the validation and verification of modeling and simulation tools developed for advanced cryogenic upper stage technologies. Continue development of hydrocarbon engine components for integration and					

PE 0603216F: Aerospace Propulsion and Power Technology
Air Force

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			D	ATE: Febru	ary 2012	
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 3: Advanced Technology Development (ATD)	R-1 ITEM NOMENCLATURE PE 0603216F: Aerospace Propulsion at Power Technology		ROJECT 4922: Spac	e & Missile	Rocket Pro	pulsion
B. Accomplishments/Planned Programs (\$ in Millions)		FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
demonstration in an advanced hydrocarbon engine concept for futu scale preburner and turbine component testing to demonstrate hydrocich material manufacturing scale-up effort to support hydrocarbon component scale-up and characterization for advanced hydrocarbon kerosene. Note: In FY 2012, funding is decreased due to higher Air	rocarbon boost technologies. Continue ox- boost demonstration program. Conduct n engine technologies using fuels other than					
FY 2013 Base Plans: Continue development of hydrocarbon engine components for integration hydrocarbon engine concept for future reusable launch vehicles. Component testing to demonstrate hydrocarbon boost technologies, scale-up effort to support hydrocarbon boost demonstration program	ontinue sub-scale preburner and turbine Complete ox-rich material manufacturing					
FY 2013 OCO Plans: N/A						
Title: Major Thrust 2.		3.696	3.748	-	-	-
Description: Develop solar electric, electric, and monopropellant postellites, upper stages, orbit transfer vehicles, and satellite maneuvers.						
FY 2011 Accomplishments: Initiated scale-up of micro propulsion technologies for spacecraft wi Continued hardware scale-up and prepared to conduct testing of ha thrust or high efficiency) propulsion system for satellites. Scaled-up spacecraft propulsion systems.	rdware for an advanced multi-mode (high					
FY 2012 Plans: Conduct scale-up of propulsion technologies for spacecraft with the hardware scale-up and conduct testing of hardware for an advance propulsion system for satellites. Build components for integration are chemical thrusters for spacecraft propulsion systems.	d multi-mode (high thrust or high efficiency)					
FY 2013 Base Plans: Decrease in FY 2013 due to higher Department of Defense prioritie	s.					
FY 2013 OCO Plans:						

PE 0603216F: Aerospace Propulsion and Power Technology Air Force

Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			D	ATE: Febru	ary 2012			
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 3: Advanced Technology Development (ATD)	PROJECT 634922: Space & Missile Rocket Prop							
B. Accomplishments/Planned Programs (\$ in Millions)		FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total		
N/A								
Title: Major Thrust 3.		2.409	2.017	1.587	-	1.587		
Description: Develop and demonstrate missile propulsion and Poballistic missiles.	st Boost Control Systems technologies for							
FY 2011 Accomplishments: Continued development of advanced missile propulsion technological developments providing sub-scale validation of modeling and simulation.	•							
FY 2012 Plans: Continue development of advanced missile case, insulation, and n component developments providing sub-scale validation of modelic component development and transition into next generation integral.	ng and simulation tools. Complete propellant							
FY 2013 Base Plans: Continue development of advanced missile case, insulation, and n component developments providing sub-scale validation of modelidue to higher Department of Defense priorities.								
FY 2013 OCO Plans: N/A								
Title: Major Thrust 4.		1.042	0.971	1.179	-	1.179		
Description: Develop and demonstrate aging and surveillance ted lifetime prediction uncertainty for individual motors, enabling motor								
FY 2011 Accomplishments: Continued integration and full-scale demonstration of advanced agrocket motors to validate and verify modeling and simulation tools assessment of effort modeling critical defects in solid rocket motors.	and component technologies. Completed							
FY 2012 Plans:								

PE 0603216F: Aerospace Propulsion and Power Technology Air Force

Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force

APPROPRIATION/BUDGET ACTIVITY

3600: Research, Development, Test & Evaluation, Air Force
BA 3: Advanced Technology Development (ATD)

PATE: February 2012

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PE 0603216F: Aerospace Propulsion and Power Technology

B. Accomplishments/Planned Programs (\$ in Millions) Continue integration and full-scale demonstration of advanced aging and surveillance tools for solid rocket	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
motors to validate and verify modeling and simulation tools and component technologies.					
FY 2013 Base Plans: Continue integration and full-scale demonstration of advanced aging and surveillance tools for solid rocket motors to validate and verify modeling and simulation tools and component technologies.					
FY 2013 OCO Plans: N/A					
Accomplishments/Planned Programs Subtotals	29.357	27.596	22.446	-	22.446

C. Other Program Funding Summary (\$ in Millions)

			FY 2013	FY 2013	FY 2013					Cost To	
<u>Line Item</u>	FY 2011	FY 2012	Base	OCO	<u>Total</u>	FY 2014	FY 2015	FY 2016	FY 2017	Complete	Total Cost
• N/A: <i>N/A</i>	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	Continuing	Continuing

D. Acquisition Strategy

N/A

E. Performance Metrics

Please refer to the Performance Base Budget Overview Book for information on how Air Force resources are applied and how those resources are contributing to Air Force performance goals and most importantly, how they contribute to our mission.

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Exhibit R-2A, RDT&E Project Just	ification: PE	3 2013 Air Fo	orce						DATE: Febr	uary 2012	
APPROPRIATION/BUDGET ACTIV 3600: Research, Development, Test BA 3: Advanced Technology Develo		R-1 ITEM N PE 0603216 Power Tech	6F: <i>Aerospa</i>		n and	PROJECT 635098: Advanced Aerospace Propulsion					
COST (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	FY 2014	FY 2015	FY 2016	FY 2017	Cost To Complete	Total Cost
635098: Advanced Aerospace Propulsion	12.744	30.117	9.553	-	9.553	18.811	42.427	39.140	29.523	Continuing	Continuing

Note

Note: In FY 2012, funding in this project is increased to complete scramjet engine flight demonstrations.

A. Mission Description and Budget Item Justification

This project develops and demonstrates, via ground and flight tests, the scramjet propulsion cycle to a technology readiness level appropriate for full integration with other engine cycles (including turbine and rocket-based) to provide the Air Force with transformational military capabilities. The primary focus is on the hydrocarbonfueled, scramjet engine. Multi-cycle engines will provide the propulsion systems for possible application to support aircraft and weapon platforms operating over the range of Mach 0 to 8+. Efforts include scramjet flow-path optimization to enable operation over the widest possible range of Mach numbers, active combustion control to assure continuous positive thrust (even during mode transition), robust flame-holding to maintain stability through flow distortions, and maximized volume-to-surface area to minimize the thermal load imposed by the high-speed engine. Thermal management plays a vital role in scramjet and combined cycle engines, including considerations for protecting low speed propulsion systems (e.g., turbine engines) during hypersonic flight.

B. Accomplishments/Planned Programs (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
Title: Major Thrust 1.	12.744	30.117	9.553	_	9.553
Description: Develop and demonstrate technologies for a hydrocarbon-fueled scramjet with robust operation over a range of Mach 4 to 8.					
FY 2011 Accomplishments: Continued flight testing of a scramjet engine demonstrator. Analyzed flight test data and began preparing a final report. Demonstrated small- scale scramjet engine to Technology Readiness Level 6.					
FY 2012 Plans: Complete flight testing of a scramjet engine demonstrator. Analyze flight test data and complete a final report. Develop and demonstrate tactically compliant subsystems, including scramjet engine start system, fuel system, and engine controls. Note: In FY 2012, efforts in this thrust are increased to complete scramjet engine flight demonstrations.					
FY 2013 Base Plans: Continue development and demonstration of tactically compliant subsystems, including scramjet engine start system, fuel system, and engine controls. Complete component demonstration of tactically compliant cold start					

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force	DATE: February 2012		
APPROPRIATION/BUDGET ACTIVITY	R-1 ITEM NOMENCLATURE	PROJECT	
3600: Research, Development, Test & Evaluation, Air Force	PE 0603216F: Aerospace Propulsion and	635098: Ad	dvanced Aerospace Propulsion
BA 3: Advanced Technology Development (ATD)	Power Technology		

B. Accomplishments/Planned Programs (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
system. Initiate design of flight weight scramjet engine cold start system, fuel system components, and advanced engine control system. Decrease in FY 2013 due to higher Department of Defense priorities.					
FY 2013 OCO Plans:					
N/A					
Accomplishments/Planned Programs Subtotals	12.744	30.117	9.553	-	9.553

C. Other Program Funding Summary (\$ in Millions)

			FY 2013	FY 2013	FY 2013					Cost To	
<u>Line Item</u>	FY 2011	FY 2012	Base	OCO	<u>Total</u>	FY 2014	FY 2015	FY 2016	FY 2017	Complete	Total Cost
• N/A: <i>N/A</i>	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	Continuing	Continuing

D. Acquisition Strategy

N/A

E. Performance Metrics

Please refer to the Performance Base Budget Overview Book for information on how Air Force resources are applied and how those resources are contributing to Air Force performance goals and most importantly, how they contribute to our mission.

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force									DATE: February 2012			
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 3: Advanced Technology Development (ATD)				R-1 ITEM N PE 0603216 Power Tech	6F: Aerospa	_	n and	PROJECT 63681B: Advanced Turbine Engine Gas Generator			Gas	
COST (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	FY 2014	FY 2015	FY 2016	FY 2017	Cost To Complete	Total Cost	
63681B: Advanced Turbine Engine Gas Generator	33.646	32.988	34.789	-	34.789	32.301	18.612	18.695	18.663	Continuing	Continuing	

A. Mission Description and Budget Item Justification

This project develops and demonstrates technology to increase turbine engine operational reliability, durability, mission flexibility, and performance while reducing weight, fuel consumption, and cost of ownership. The objective is to provide the continued evolution of technologies into an advanced gas generator in which the performance, cost, durability, reparability, and maintainability can be assessed in a realistic engine environment. The gas generator, or core, is the basic building block of the engine and nominally consists of a compressor, a combustor, a high-pressure turbine, mechanical systems, and core subsystems. Experimental core engine demonstration validates engineering design tools and enhances rapid, low-risk transition of key engine technologies into engineering development, where they can be applied to derivative and/or new systems. These technologies are applicable to a wide range of military and commercial systems including aircraft, missiles, land combat vehicles, ships, and responsive space launch. Component technologies are demonstrated in a core (sub-engine). This project also assesses the impact of low spool components (such as inlet systems, fans, low pressure turbines, and exhaust systems) and system level technologies (such as integrated power generators and thermal management systems) on core engine performance and durability in "core-centric engine" demonstration. The core performances of this project are validated on demonstrator engines in Project 4921 of this Program Element. A portion of this project supports the demonstration of adaptive cycle technologies, which develop component technology for an adaptive cycle engine architecture that provides optimized performance, fuel efficiency, and durability for widely varying mission needs.

B. Accomplishments/Planned Programs (\$ in Millions)			FY 2013	FY 2013	FY 2013
	FY 2011	FY 2012	Base	oco	Total
Title: Major Thrust 1.	20.290	19.790	15.322	-	15.322
Description: Design, fabricate, and demonstrate performance predictions in core engines, using innovative engine cycles and advanced materials for turbofan/turbojet engines.					
FY 2011 Accomplishments: Continued hardware fabrication and initiate assembly of high temperature capable, durable compressor, combustor, and turbine for adaptive core engine. Completed detailed design and initiate fabrication of component technologies for a core-centric durability engine demonstration. Conducted fabrication of component technologies for increased reliability, maintainability, and affordability for potential transition to fielded systems. Conducted preliminary design and initiate detailed design of system-level technologies and weapon systems integration on core engine performance.					
FY 2012 Plans: Complete hardware fabrication, assembly and experimental demonstration of high temperature capable, durable compressor, combustor, and turbine for adaptive core engine. Continue fabrication of component					

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			D	ATE: Febru	ary 2012		
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 3: Advanced Technology Development (ATD)	R-1 ITEM NOMENCLATURE PE 0603216F: Aerospace Propulsion an Power Technology	PROJECT nd 63681B: Advanced Turbine Engine Gas Generator					
B. Accomplishments/Planned Programs (\$ in Millions)		FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	
technologies and initiate assembly for a core-centric durability engine component technologies for increased reliability, maintainability, and a systems. Conduct detailed design of system-level technologies and we performance.							
FY 2013 Base Plans: Evaluate and conduct post demonstration assessment of high temperators combustor, and turbine for adaptive core engine.	ature capable, durable compressor,						
FY 2013 OCO Plans: N/A							
Title: Major Thrust 2.		13.356	13.198	3.500	-	3.500	
Description: Design, fabricate, and demonstrate high overall pressure and affordability with lower fuel consumption for turbofan/turboshaft er							
FY 2011 Accomplishments: Completed preliminary design of core for efficient core engine concept including high efficiency, high pressure ratio, high temperature capability release combustor, and high work, high cooling effectiveness turbine to system and advanced mechanical systems. Continued selective risk for RPA small versatile affordable advanced core engine. Continued defficient small engine component technologies including high efficience capability compressor, high efficiency, high heat release combustor, a uncooled turbine for use in RPA applications.	lity compressor, high efficiency, high heat with an integrated thermal management reduction experimental demonstrations etailed design and initiate fabrication of cy, high pressure ratio, high temperature						
FY 2012 Plans: Initiate detailed design, fabrication and begin assembly and instrument concept with advanced core technologies including high efficiency, high capability compressor, high efficiency, high heat release combustor, a turbine with an integrated thermal management system and advanced risk reduction experimental demonstrations of RPA small versatile affordabrication of efficient small engine component technologies including temperature capability compressor, high efficiency, high heat release to the component technologies.	gh pressure ratio, high temperature and high work, high cooling effectiveness dimechanical systems. Continue selective ordable advanced core engine. Continue high efficiency, high pressure ratio, high						

PE 0603216F: Aerospace Propulsion and Power Technology Air Force

Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force		DATE: February 2012					
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 3: Advanced Technology Development (ATD)	nd 6	PROJECT 33681B: Adva Generator	Advanced Turbine Engine Gas				
B. Accomplishments/Planned Programs (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total		
effectiveness or uncooled turbine for use in RPA applications. Initiate efficient and very high pressure ratio core engine.	conceptual design for advanced very						
FY 2013 Base Plans: Continue detailed design, fabrication and begin assembly and instrum concept with advanced core technologies including high efficiency, high capability compressor, high efficiency, high heat release combustor, a turbine with an integrated thermal management system and advanced 2013 due to higher Department of Defense priorities.							
FY 2013 OCO Plans: N/A							
Title: Major Thrust 3.		-	-	15.967	-	15.967	
Description: Design, fabricate, and demonstrate performance, durab adaptive turbine engine core technologies.	ility, and operability technologies to mature						
FY 2011 Accomplishments: N/A							
FY 2012 Plans: N/A							
FY 2013 Base Plans: Conduct design of core technologies for application to adaptive turbin consumption, improved thrust-to-weight, and reduced cost. Perform p analysis and assessment.							
FY 2013 OCO Plans: N/A							
Accom	plishments/Planned Programs Subtotals	33.64	6 32.988	34.789	-	34.789	

PE 0603216F: Aerospace Propulsion and Power Technology Air Force

Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force	DATE: February 2012		
APPROPRIATION/BUDGET ACTIVITY	R-1 ITEM NOMENCLATURE	PROJECT	
3600: Research, Development, Test & Evaluation, Air Force	PE 0603216F: Aerospace Propulsion and	63681B: Ad	Ivanced Turbine Engine Gas
BA 3: Advanced Technology Development (ATD)	Power Technology	Generator	

C. Other Program Funding Summary (\$ in Millions)

			FY 2013	FY 2013	FY 2013					Cost To	
<u>Line Item</u>	FY 2011	FY 2012	Base	OCO	<u>Total</u>	FY 2014	FY 2015	FY 2016	FY 2017	Complete	Total Cost
• N/A: <i>N/A</i>	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	Continuing	Continuing

D. Acquisition Strategy

N/A

E. Performance Metrics

Please refer to the Performance Base Budget Overview Book for information on how Air Force resources are applied and how those resources are contributing to Air Force performance goals and most importantly, how they contribute to our mission.

PE 0603216F: Aerospace Propulsion and Power Technology Air Force

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