Exhibit R-2, RDT&E Budget Item Justification: PB 2013 Air Force

R-1 ITEM NOMENCLATURE

3600: Research, Development, Test & Evaluation, Air Force

PE 0602203F: Aerospace Propulsion

BA 2: Applied Research

APPROPRIATION/BUDGET ACTIVITY

COST (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	FY 2014	FY 2015	FY 2016	FY 2017	Cost To Complete	Total Cost
Total Program Element	198.878	207.406	232.547	-	232.547	200.918	165.900	168.511	177.525	Continuing	Continuing
623012: Advanced Propulsion Technology	22.074	20.367	23.637	-	23.637	23.345	22.160	24.276	27.254	Continuing	Continuing
623048: Combustion and Mechanical Systems	17.734	20.069	15.874	-	15.874	13.886	12.744	12.829	13.103	Continuing	Continuing
623066: Turbine Engine Technology	64.278	67.702	102.188	-	102.188	75.523	42.355	42.628	43.520	Continuing	Continuing
623145: Aerospace Power Technology	31.346	32.639	30.061	-	30.061	27.801	28.677	28.739	28.848	Continuing	Continuing
624847: Rocket Propulsion Technology	56.966	60.390	55.293	-	55.293	54.888	54.689	54.727	59.374	Continuing	Continuing
625330: Aerospace Fuel Technology	6.480	6.239	5.494	-	5.494	5.475	5.275	5.312	5.426	Continuing	Continuing

## A. Mission Description and Budget Item Justification

This program develops propulsion and power technologies to achieve enabling and revolutionary aerospace technology capabilities. The program has six projects, each focusing on a technology area critical to the Air Force. The Advanced Propulsion Technology project develops high-speed air breathing propulsion engines to include combined cycle, ramjet, and hypersonic scramjet technologies to enable revolutionary propulsion capability for the Air Force. The Combustion and Mechanical Systems project evaluates lubricants and combustion concepts and technologies for new and existing engines. The Turbine Engine Technology project develops enabling capabilities to enhance performance and affordability of existing weapon systems and develops component technologies for ultra high pressure ratio, substantially improved durability, and adaptive cycle engine architecture to provide optimized performance, fuel efficiency, and life for widely varying mission needs. The Aerospace Power Technology project develops electrical power and thermal management technologies for military applications that are part of energy optimized aircraft development. The Rocket Propulsion Technology project develops advances in rocket propulsion technologies for space access, space maneuver, missiles, the sustainment of strategic systems, and tactical rockets. The Aerospace Fuel Technology project evaluates hydrocarbon-based fuels for legacy and advanced turbine engines, scramjets, pulse detonation, and combined-cycle engines. Efforts in this program have been coordinated through the Reliance 21 process to harmonize efforts and eliminate duplication. This program is in Budget Activity 2, Applied Research, since it develops and determines the technical feasibility and military utility of evolutionary and revolutionary technologies.

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**DATE:** February 2012

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APPROPRIATION/BUDGET ACTIVITY

**R-1 ITEM NOMENCLATURE** 

3600: Research, Development, Test & Evaluation, Air Force

PE 0602203F: Aerospace Propulsion

BA 2: Applied Research

B. Program Change Summary (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
Previous President's Budget	207.049	207.508	204.967	-	204.967
Current President's Budget	198.878	207.406	232.547	-	232.547
Total Adjustments	-8.171	-0.102	27.580	-	27.580
<ul> <li>Congressional General Reductions</li> </ul>	-	-0.102			
<ul> <li>Congressional Directed Reductions</li> </ul>	-	-			
Congressional Rescissions	-	-			
Congressional Adds	-	-			
<ul> <li>Congressional Directed Transfers</li> </ul>	-	-			
Reprogrammings	-	-			
SBIR/STTR Transfer	-3.474	-			
<ul> <li>Other Adjustments</li> </ul>	-4.697	-	27.580	-	27.580

## **Change Summary Explanation**

FY11: Other Adjustments include -1.697 Congressional General Reductions and -3.000 Congressional Directed Reductions

FY13: Increase due to higher Air Force priorities in Turbine Engine Technology

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force										DATE: February 2012		
APPROPRIATION/BUDGET ACTIV 3600: Research, Development, Test BA 2: Applied Research		I <b>OMENCLA</b> 1 3F: <i>Aerospad</i>		n	PROJECT 623012: Advanced Propulsion Technology							
COST (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	FY 2014	FY 2015	FY 2016	FY 2017	Cost To Complete	Total Cost	
623012: Advanced Propulsion Technology	22.074	20.367	23.637	-	23.637	23.345	22.160	24.276	27.254	Continuing	Continuing	

### A. Mission Description and Budget Item Justification

B. Accomplishments/Planned Programs (\$ in Millions)

This project develops combined/advanced cycle air breathing high-speed (up to Mach 4) and hypersonic (Mach 4 to 8+) propulsion technologies to provide revolutionary propulsion options for the Air Force. These new engine technologies will enable future high-speed/hypersonic weapons and aircraft concepts. The primary focus is on hydrocarbon-fueled engines capable of operating over a broad range of flight Mach numbers. Efforts include modeling, simulations, and proof of concept demonstrations of critical components; advanced component development; and ground-based demonstrations.

2. Accomplishments/Figurite (# III Millions)	FY 2011	FY 2012	Base	OCO	Total
Title: Major Thrust 1.	2.565	1.650	1.650	-	1.650
<b>Description:</b> Develop advanced fuel-cooled scramjet engine technologies to support flight demonstration and enable the broad application of hypersonics to meet future warfighter needs.					
FY 2011 Accomplishments:  Developed and demonstrated flight weight engine components and advanced engine control logic. Assessed advanced instrumentation with control logic to improve scramjet operability. Performed trajectory optimization for flight test. Conducted ground test of advanced scramjet start technique. Completed fabrication of flight test hardware to demonstrate ramjet to scramjet transition.					
FY 2012 Plans:  Develop and demonstrate advanced engine control systems and flight weight scramjet engine components.  Develop and demonstrate closed loop engine control system with advanced instrumentation to increase scramjet engine operability at low scramjet Mach numbers. Conduct flight test using sounding rocket launch to demonstrate transition from ramjet to scramjet.					
FY 2013 Base Plans: Continue development and demonstration of advanced engine control systems and flight weight scramjet engine components. Based on prior ground and flight testing, refine and demonstrate closed loop engine control system with advanced instrumentation to increase scramjet engine operability at low scramjet Mach numbers. Conduct direct connect testing of flight weight scramjet components for cold start systems.					
FY 2013 OCO Plans:					

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FY 2013 | FY 2013 | FY 2013

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			D	ATE: Febru	ary 2012	
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research	R-1 ITEM NOMENCLATURE PE 0602203F: Aerospace Propulsion	project 623012: Advanced Propulsion Technol				
B. Accomplishments/Planned Programs (\$ in Millions)		FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
N/A						
<i>Title:</i> Major Thrust 2.		0.165	0.165	0.120	-	0.120
<b>Description:</b> Conduct assessments, technology design trades, and engines (CCEs) and air breathing hypersonic propulsion technologic						
FY 2011 Accomplishments: Conducted further trade studies to determine military payoff and est Defined component and engine performance objectives to enable dedemonstrators jointly with National Aeronautics and Space Administ Research Projects Agency (DARPA). Developed technology matura for advanced components for turbine-based and rocket-based CCEs	evelopment of affordable hypersonic flight tration (NASA) and Defense Advanced tion plan, including test facility requirements,					
FY 2012 Plans: Continue to conduct trade studies to determine military payoff and e Improve definition of component and engine performance objectives hypersonic flight demonstrators jointly with NASA and DARPA. Upd test facility requirements, for advanced components for turbine-base	s to enable development of affordable ate technology maturation plan, including					
FY 2013 Base Plans: Continue to conduct trade studies to determine military payoff and elemprove definition of component and engine performance objectives hypersonic flight demonstrators jointly with NASA and DARPA. Upd test facility requirements, for advanced components for turbine-base component development roadmapping.	s to enable development of affordable ate technology maturation plan, including					
FY 2013 OCO Plans: N/A						
Title: Major Thrust 3.		19.344	18.552	21.867	-	21.867
<b>Description:</b> Develop robust hydrocarbon fueled scramjet engine coperformance, operability, durability, and scalability for future platform						
FY 2011 Accomplishments:						

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APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force 3A 2: Applied Research  B. Accomplishments/Planned Programs (\$ in Millions)  Developed advanced engine components to improve scramjet of laws for reusable applications. Developed techniques to decrease 3.5 to provide robust options for CCEs. Developed low internal of engine components. Ground tested subscale components/comb scramjet engines.  FY 2012 Plans:  Develop advanced engine components to improve scramjet operates for reusable applications. Develop techniques to decrease to provide robust options for CCEs. Develop low internal drag flat components. Design and initiate fabrication of heavy weight scramiet engines.	operating management to a scramjet to a scrampet to a scra	nargin and to et take-over f e stabilization represent me rgin and to re cake-over froi ization device	o refine scra from Mach 4 devices and edium scale efine scramj m Mach 4.5 es and flight	mjet scaling 5.5 to Mach d flight test (5 to 20 times) et scaling to Mach 3.5	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
Developed advanced engine components to improve scramjet of laws for reusable applications. Developed techniques to decrease 3.5 to provide robust options for CCEs. Developed low internal components components. Ground tested subscale components/combiscramjet engines.  FY 2012 Plans:  Develop advanced engine components to improve scramjet oper laws for reusable applications. Develop techniques to decrease to provide robust options for CCEs. Develop low internal drag flat components. Design and initiate fabrication of heavy weight scra	se scramje drag flame oustors to r erating mar scramjet to ame stabili	et take-over for take stabilization represent me regin and to recake-over from the restate of th	from Mach 4 devices and edium scale efine scramjo m Mach 4.5 es and flight	et scaling to Mach 3.5 to test 3.5 to Mach 3.5 test engine		FY 2012	1		
laws for reusable applications. Developed techniques to decrease 3.5 to provide robust options for CCEs. Developed low internal companies components. Ground tested subscale components/combines engines.  FY 2012 Plans:  Develop advanced engine components to improve scramjet operates for reusable applications. Develop techniques to decrease to provide robust options for CCEs. Develop low internal drag flat components. Design and initiate fabrication of heavy weight scrape.	se scramje drag flame oustors to r erating mar scramjet to ame stabili	et take-over for take stabilization represent me regin and to recake-over from the restate of th	from Mach 4 devices and edium scale efine scramjo m Mach 4.5 es and flight	et scaling to Mach 3.5 to test 3.5 to Mach 3.5 test engine					
Develop advanced engine components to improve scramjet ope laws for reusable applications. Develop techniques to decrease to provide robust options for CCEs. Develop low internal drag fla components. Design and initiate fabrication of heavy weight scra	scramjet ta ame stabili	ake-over from	m Mach 4.5 es and flight	to Mach 3.5 test engine					
zoranijot originoo.					1				
FY 2013 Base Plans: Continue to develop advanced engine components to improve secramjet scaling laws for reusable applications. Continue to develover from Mach 4.5 to Mach 3.5 to provide robust options for CC flame stabilization devices and flight test engine components. Combustor in medium scale (5 to 20 times) scramjet engines. Initiation 20 times) scramjet engines operating at Mach 3.5 to Mach 7 cores.	velop techn CEs. Contii Complete fa nitiate direct	niques to dec nue to devel abrication of	rease scran op low inter heavy weigh	njet take- nal drag nt scramjet					
FY 2013 OCO Plans: N/A									
A	ccomplish	hments/Plar	nned Progra	ams Subtotals	22.074	4 20.367	23.637	-	23.63
C. Other Program Funding Summary (\$ in Millions)	FY 2013	EV 2042	EV 2042					Cost To	
<u>Line Item</u> <u>FY 2011</u> <u>FY 2012</u> • N/A: N/A 0.000 0.000	Base 0.000	FY 2013 OCO 0.000	FY 2013 Total 0.000	<b>FY 2014</b> 0.000	<b>FY 2015</b> 0.000	<b>FY 2016</b> 0.000		Complete Continuing	
D. Acquisition Strategy N/A									

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force		DATE: February 2012
APPROPRIATION/BUDGET ACTIVITY	R-1 ITEM NOMENCLATURE	PROJECT
3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research	PE 0602203F: Aerospace Propulsion	623012: Advanced Propulsion Technology
E. Performance Metrics  Please refer to the Performance Base Budget Overview Book for Force performance goals and most importantly, how they contribute the performance goals are most importantly.		ed and how those resources are contributing to Air

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force										DATE: February 2012		
APPROPRIATION/BUDGET ACTIV 3600: Research, Development, Tes BA 2: Applied Research		I <b>OMENCLA</b> 3F: <i>Aerospa</i>		n	PROJECT 623048: Combustion and Mechanical Systems							
COST (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	FY 2014	FY 2015	FY 2016	FY 2017	Cost To Complete	Total Cost	
623048: Combustion and Mechanical Systems	17.734	20.069	15.874	-	15.874	13.886	12.744	12.829	13.103	Continuing	Continuing	

### A. Mission Description and Budget Item Justification

This project evaluates lubricants, mechanical systems, and combustion concepts for advanced turbine engines, pulse detonation engines, and combined cycle engines. This project also develops technologies to increase turbine engine operational reliability, durability, mission flexibility, maintainability, and performance while reducing weight, fuel consumption, and cost of ownership. Applications include missiles, aircraft, and sustained high-speed vehicles. Analytical and experimental areas of emphasis include lubricants, bearings, mechanical systems diagnostics, mechanical systems prognostics, rotordynamics, oil-less engine technology, optical diagnostics, fundamental combustion, detonations, combustors, and afterburners. Lubricants for these engines must be thermally stable, cost-effective, and operate over a broad range of conditions. Advanced combustion concepts must be cost-effective, durable, and reduce pollutant emissions. A portion of this project supports adaptive cycle technologies. This effort develops component technology for an adaptive cycle engine architecture that provides optimized performance/fuel efficiency for widely varying mission needs.

B. Accomplishments/Planned Programs (\$ in Millions)			FY 2013	FY 2013	FY 2013
	FY 2011	FY 2012	Base	oco	Total
Title: Major Thrust 1.	7.541	8.719	6.202	-	6.202
<b>Description:</b> Develop, test, and evaluate revolutionary combustion and propulsion concepts for gas turbine, pulse detonation, and combined cycle engines for missiles, manned and unmanned systems.					
FY 2011 Accomplishments:  Tested full-scale inter-turbine burner (ITB) concepts at relevant engine conditions. Investigated novel valving concepts for pulse detonation engines. Studied pulse detonation engine-turbine interactions. Explored the use of regenerative fuel cooling with pulse detonation engines and other combustion systems. Demonstrated novel small internal combustion engine concepts that improve system performance. Used advanced modeling and simulation tools to understand combustion processes and to guide combustion system design. Employed new chemistry models for alternative fuels. Tested concept designs for adaptive combustors for ultra efficient turbine					
engine components which reduce harmful emissions.  FY 2012 Plans:  Evaluate alternative fuels in combustion systems at relevant engine conditions. Test full-scale compact					
combustor concept relevant to highly efficient, embedded turbine engine goals. Demonstrate small-scale propulsion system operation using reduced-octane fuels. Employ new physical models in simulation tools. Investigate pressure gain combustion concepts for application to propulsion systems. Continue studies of pulse					

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EV 2042 EV 2042 EV 2042

Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			D	ATE: Febru	ary 2012	
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research	R-1 ITEM NOMENCLATURE PE 0602203F: Aerospace Propulsion		PROJECT 623048: Com	bustion and	Mechanica	l Systems
B. Accomplishments/Planned Programs (\$ in Millions)		FY 201	1 FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
detonation engine-turbine interactions. Investigate feasibility of rota detonation engines.	ry detonation engines (RDE) and continuous					
FY 2013 Base Plans: Develop new models for combustion processes at high pressure contract that produce low pollutant emissions. Test RDE concepts. Decrease Defense priorities.						
FY 2013 OCO Plans: N/A						
Title: Major Thrust 2.		1.17	71 1.311	1.128	-	1.128
<b>Description:</b> Develop and demonstrate optical, electromechanical, application to revolutionary propulsion technologies.	, and laser diagnostic tools and sensors for					
FY 2011 Accomplishments:  Used two-color planar laser induced fluorescence techniques to me combustion systems. Developed robust line-of-sight measurement and apply to engine systems. Developed simultaneous high-speed particle-image velocimetry for measurements of species and velocit Expanded line-of-sight measurement techniques for temperature at lines of sight and tomographic reconstruction of complex reacting fl Applied advanced optical diagnostics suites for characterization and afterburners.	techniques for temperature and species planar laser-induced fluorescence and ty fields in practical combustion devices. nd species to include many simultaneous owfields characteristic of real-world hardware.					
FY 2012 Plans: Apply line-of-sight measurement techniques for temperature and spengine environment. Demonstrate simultaneous high-speed planar image velocimetry for measurements of species and velocity fields tomographic reconstruction of reacting flowfields in relevant combu	laser-induced fluorescence and particle- in practical combustion devices. Demonstrate					
FY 2013 Base Plans:						
			·	•	•	

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			D	ATE: Febru	ary 2012		
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research	R-1 ITEM NOMENCLATURE PE 0602203F: Aerospace Propulsion	ion PROJECT 623048: Combustion and Mechanical					
B. Accomplishments/Planned Programs (\$ in Millions)		FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	
Apply advanced laser diagnostics and novel optics configurations to Demonstrate particle image velocimetry in high pressure combustic measurement techniques for combustion temperature and species	on test apparatus. Investigate high-speed						
FY 2013 OCO Plans: N/A							
Title: Major Thrust 3.		4.463	4.966	4.181	-	4.18	
<b>Description:</b> Develop, test, and qualify advanced turbine engine luspecifications for aviation engine lubricants.	bricants. Generate and maintain military						
FY 2011 Accomplishments:  Completed Technology Readniness Level (TRL) 5 full-scale bearin candidate oil in preparation for an engine demonstration. Supporte operational fleet by coordinating with engine manufacturers and us efficiency risk mitigation bearing and gear rig tests with enhanced engine test. Demonstrated anti-coke surface modifiers on sub-scal Expanded development of intelligent prognostics for lubrication systube system thermal and health management technologies for high	d full transition of enhanced ester oil to the ers. Conducted adaptive components for high ester oil in preparation for 2013 demonstration e supersonic lube system components. Stem health monitoring. Investigated advanced						
FY 2012 Plans: Demonstrate anti-coke surface modifiers on full-scale lubrication symechanical system health management control algorithms for active technologies for intelligent lube system prognostics and health more devices, real-time oil debris monitoring, and vibration sensing. Develochnologies for reduced heat generation and improved heat dissipation.	re rotor thrust balancing. Develop suite of nitoring, such as integrated debris capture elop lubrication system thermal management						
FY 2013 Base Plans:  Demonstrate lube system health management control algorithms we enhanced ester oils in demonstrator turbine engines. Continue investmanagement technologies for fuel efficient turbine engines. Developmentally.	estigating advanced lube system thermal						
FY 2013 OCO Plans:							

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Exhibit R-2A, RDT&E Project Justification: F	PB 2013 Air Fo	orce					D	ATE: Febru	uary 2012		
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluati BA 2: Applied Research	on, Air Force		R-1 ITEM NO PE 06022031			PROJECT 623048: Combustion and Mechanical Sy					
B. Accomplishments/Planned Programs (\$ i	n Millions)					FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	
N/A											
Title: Major Thrust 4.						4.559	5.073	4.363	-	4.363	
<b>Description:</b> Develop and test advanced bear intermediate, and large-sized turbine engine as		chnology and	d bearing cor	ncepts for sr	nall,						
FY 2011 Accomplishments: Investigated fatigue life and spall propagation of VAR) bearings. Completed mechanical system efficiency. Developed coupled bearing and rote for advanced engines. Continued developing relimited-life engines. Completed fabrication of an	s risk mitigation or dynamic mo eliable bearing	on test activiti dels for virtua technologies	ies for adapti al simulation s for sustaine	ve compone of mechanied ad hi-mach r	ents for high cal systems						
FY 2012 Plans: Conduct shakedown tests of active thrust balar devices for highly loaded engine thrust bearing incorporate into thrust load control algorithm. Esensing on seeded fault bearing rig tests. Deve with full-scale bearing experimental performance.	s. Develop be emonstrate oi elop new beari	aring spall de I debris moni	ebris monitor itoring techno	ing model a ology fused	nd limits and with vibration						
FY 2013 Base Plans: Conduct parametric active thrust control experi bearing tests to validate reliable active and aut vibration, and oil debris sensing for complete T plans for demonstrating active thrust control sy	onomous thrus RL 5 mechani	st load contro cal system h	ol. Integrate a ealth manag	active thrust ement syste	control,						
<b>FY 2013 OCO Plans:</b> N/A											
		Accomplis	hments/Plar	nned Progr	ams Subtotals	17.734	20.069	15.874	-	15.874	
C. Other Program Funding Summary (\$ in M	illions)	FY 2013	FY 2013	FY 2013					Cost To		
Line Item FY 201	1 FY 2012	FY 2013 Base	OCO	Total	FY 2014	FY 2015	FY 2016	FY 2017		Total Cost	
• N/A: <i>N/A</i> 0.00		0.000	0.000	0.000	0.000	0.000	0.000			Continuing	

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			DATE: February 2012
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research	R-1 ITEM NOMENCLATURE PE 0602203F: Aerospace Propulsion	<b>PROJECT</b> 623048: <i>Co</i>	ombustion and Mechanical Systems
D. Acquisition Strategy N/A			

## E. Performance Metrics

Please refer to the Performance Base Budget Overview Book for information on how Air Force resources are applied and how those resources are contributing to Air Force performance goals and most importantly, how they contribute to our mission.

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Exhibit R-2A, RDT&E Project Just	ification: PE	3 2013 Air Fo	orce						DATE: Febr	uary 2012	
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research					I <b>OMENCLA</b> 3F: <i>Aerospa</i>		n	PROJECT 623066: Turbine Engine Technology			
COST (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	FY 2014	FY 2015	FY 2016	FY 2017	Cost To Complete	Total Cost
623066: Turbine Engine Technology	64.278	67.702	102.188	-	102.188	75.523	42.355	42.628	43.520	Continuing	Continuing

### A. Mission Description and Budget Item Justification

This project develops technology to increase turbine engine operational reliability, mission flexibility, and performance, while reducing weight, fuel consumption, and cost of ownership. Analytical and experimental areas of emphasis are fans and compressors, high temperature combustors, turbines, internal flow systems, controls, augmentor and exhaust systems, integrated power and thermal management systems, engine inlet integration, mechanical systems, adaptive cycle technologies, and structural design. This project develops component technology for an adaptive cycle engine architecture that provides optimized performance/fuel efficiency for widely varying mission needs. This project supports joint Department of Defense, agency, and industry efforts to focus turbine propulsion technology on national needs. The program plan is relevant across capability areas for global responsive strike, capable unmanned war-fighting, tactical and global mobility, responsive space lift, and persistent intelligence, surveillance, and reconnaissance (ISR).

B. Accomplishments/Planned Programs (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
Title: Major Thrust 1.	38.101	41.133	34.578	-	34.578
<b>Description:</b> Develop core turbofan/turbojet engine components (i.e., compressors, combustors, and turbines) for fighters, bombers, sustained supersonic/hypersonic cruise vehicles, and transports.					
FY 2011 Accomplishments:  Developed and applied advanced modeling and simulation rules and tools for advanced components. Developed computational fluid dynamics methodology for analyzing turbine flows. Developed ceramic matrix composite lifing models. Conducted bench and rig tests for validation of components with significantly improved efficiency. Performed rig testing of lightweight, simple, adaptive cycle features, an efficient, wide-flow range compressor, an efficient, high temperature turbine capable of operating over large swings in required work, and an efficient, lightweight, low observable (LO)-compatible exhaust system. Developed and applied advanced modeling and simulation rules and tools to initiate definition and design of efficient, very high pressure ratio core component technologies that will offer a step change improvement in engine specific fuel consumption.					
FY 2012 Plans: Develop modeling and simulation rules and tools for advanced components including advanced interactive cost analysis tools for adaptive core components and unsteady aerodynamics/aeromechanics models. Conduct bench and rig tests to validate unsteady aerodynamics/aeromechanics models. Continue rig testing adaptive cycle features, an efficient compressor, an efficient turbine, and an efficient exhaust system. Continue to develop					

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			D	ATE: Febru	ary 2012		
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research		PROJECT 623066: Turbine Engine Technology					
B. Accomplishments/Planned Programs (\$ in Millions)		FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	
and apply advanced modeling and simulation rules and tools to inihigh pressure ratio core component technologies.	tiate definition and design of efficient, very						
FY 2013 Base Plans: Develop modeling and simulation tools for advanced components turbine durability design. Conduct bench and rig test using test using ages. Develop high resolution non-contact stress measurement a measurement. Demonstrate engine efficiency improvements from rig testing of high power low emission combustion. Develop improvextend engine operability and efficiency.	ing surface mapping thin film temperature systems for high frequency response active clearance and flow control. Conduct						
FY 2013 OCO Plans: N/A							
Title: Major Thrust 2.		19.237	19.510	14.672	-	14.67	
<b>Description:</b> Develop turbofan/turbojet engine components (i.e., f bombers, sustained supersonic strike and hypersonic cruise vehic							
FY 2011 Accomplishments:  Developed and applied advanced modeling and simulation rules a durable damping/erosion coating systems. Conducted rig testing of variable cycle engine concept. Conducted rig testing of advanced variable cycle engine concept. Rig tested lightweight, simple, LO-complex concept.	of advanced fan design for application to a low pressure turbine design for application to a						
FY 2012 Plans: Develop modeling and simulation rules and tools for advanced cor analysis tools for adaptive engine components; unsteady aerodyna combustion processes; and probability-based cooled turbine airfoi Conduct bench and rig tests to validate unsteady aerodynamics/ac cooled turbine airfoil high cycle fatigue prediction methods. Develor augmentor rig test capabilities. Continue rig testing of advanced fat design, and lightweight, simple, LO-compatible inlet and exhaust so	amics and aeromechanics models; augmentor I high cycle fatigue prediction methods. eromechanics models and probabilistic op and validate test protocols and improved an design, advanced low pressure turbine						
FY 2013 Base Plans:							

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APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research	<b>PF</b> 62	Technology	/			
B. Accomplishments/Planned Programs (\$ in Millions)		FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
Develop modeling and simulation tools including methods to predict Demonstrate methods to detect/predict incipient bearing damage to 2013 due to higher Department of Defense priorities.						
FY 2013 OCO Plans: N/A						
Title: Major Thrust 3.		5.309	5.400	3.993	-	3.993
<b>Description:</b> Develop limited life engine components for missile ar including long-range supersonic and hypersonic vehicles.	nd remotely piloted aircraft (RPA) applications,					
FY 2011 Accomplishments:  Developed and applied advanced modeling and simulation rules are Designed and rig tested advanced limited life components.	nd tools for advanced limited life components.					
FY 2012 Plans: Develop and apply advanced modeling and simulation rules and to variable area turbines, and integration/performance of lubeless bear increase pressure ratio by 50% in this size class with minimum efficiency.	rings. Develop and evaluate components to					
FY 2013 Base Plans: Develop and apply advanced modeling and simulation tools for var cooling concepts, compact augmentors, and composite structures.						
FY 2013 OCO Plans: N/A						
Title: Major Thrust 4.		1.631	1.659	1.545	-	1.545
<b>Description:</b> Develop components for turboshaft/turboprop and sn special operations aircraft, and theater transports.	nall turbofan engines for trainers, rotorcraft,					
FY 2011 Accomplishments:  Developed and applied advanced modeling and simulation rules ar	nd tools for advanced limited life components.					
FY 2012 Plans:						

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Exhibit R-2A, RDT&E Project Jus	tification: PB	2013 Air Fo	rce					D	ATE: Febr	uary 2012	
APPROPRIATION/BUDGET ACTI 3600: Research, Development, Tes BA 2: Applied Research		PROJECT 623066: Turb	ine Engine	Technology							
B. Accomplishments/Planned Pro	ograms (\$ in I	Millions)					FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
Develop and apply advanced mode detection. Develop and evaluate coconsumption, production cost, and	omponents to in	ncrease thru									
FY 2013 Base Plans: Develop and apply advanced mode gearboxes, and high performance ademonstration of engine durability	airfoils. Develo										
<b>FY 2013 OCO Plans:</b> N/A											
Title: Major Thrust 5.							-	-	47.400	) -	47.400
<b>Description:</b> Develop high perform technologies.	nance, durable	components	s which enab	ole adaptive t	turbine engi	ne					
<b>FY 2011 Accomplishments:</b> N/A											
<b>FY 2012 Plans:</b> N/A											
FY 2013 Base Plans: Conduct bench and rig tests to vali compressor, combustor, turbine, the reduce specific fuel consumption, is	ermal manage	ment, and a	ugmentor/ex	haust nozzle	• .						
<b>FY 2013 OCO Plans:</b> N/A											
			Accomplisi	nments/Plar	ned Progra	ıms Subtotal	ls 64.27	8 67.702	102.188	-	102.188
C. Other Program Funding Sumn	nary (\$ in Milli	ons)									
	<b>,</b> ,		EV 2042	EV 2042	EV 0040					Cost To	
			FY 2013	FY 2013	FY 2013					COST 10	
Line Item • N/A: N/A	FY 2011 0.000	<b>FY 2012</b> 0.000	Base 0.000	OCO 0.000	<u>Total</u> 0.000	<b>FY 2014</b> 0.000	<b>FY 2015</b> 0.000	<b>FY 2016</b> 0.000		Complete Continuing	

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force		DATE: February 2012
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research	R-1 ITEM NOMENCLATURE PE 0602203F: Aerospace Propulsion	PROJECT 623066: Turbine Engine Technology
D. Acquisition Strategy N/A		
E. Performance Metrics  Please refer to the Performance Base Budget Overview Book for Force performance goals and most importantly, how they contributed the performance goals are contributed by the performance goals and most importantly.		ed and how those resources are contributing to Air

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force										DATE: February 2012		
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research				R-1 ITEM NOMENCLATURE PE 0602203F: Aerospace Propulsion				PROJECT 623145: Aerospace Power Technology				
COST (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	FY 2014	FY 2015	FY 2016	FY 2017	Cost To Complete	Total Cost	
623145: Aerospace Power Technology	31.346	32.639	30.061	-	30.061	27.801	28.677	28.739	28.848	Continuing	Continuing	

## A. Mission Description and Budget Item Justification

This project develops electrical and thermal management technologies for military aerospace applications. Power component technologies are developed to increase reliability, maintainability, commonality, affordability, and supportability of aircraft and flight line equipment. Research is conducted in energy storage and hybrid power system technologies to enable special purpose applications. Electrical power and thermal management technologies enable future military megawatt level power and thermal management needs. This project supports development of electrical power and thermal management component and systems suitable for applications to legacy and future aircraft platforms including strike and mobility concepts. Lightweight power systems suitable for other aerospace applications are also developed.

B. Accomplishments/Planned Programs (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
Title: Major Thrust 1.	27.366	26.831	25.473	-	25.473
<b>Description:</b> Develop electrical power and thermal management component and subsystem technologies with low volume displacement for delivery of high power for manned and unmanned systems.					
FY 2011 Accomplishments:  Performed hardware-in-the-loop simulation tests to validate power and thermal management systems that provide continuous thermal balancing of critical systems over a range of mission profiles. Assessed component technologies for application to directed energy weapon concepts.					
FY 2012 Plans: Perform tip-to-tail modeling and simulation to identify solutions for platform level power and thermal management needs of next generation military air platforms.					
FY 2013 Base Plans: Design and develop adaptive power and thermal management subsystems for next generation military air platforms based on platform level tip-to-tail modeling and simulation energy optimization.					
FY 2013 OCO Plans: N/A					
Title: Major Thrust 2.	3.980	5.808	4.588	-	4.588

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force **DATE:** February 2012

APPROPRIATION/BUDGET ACTIVITY **R-1 ITEM NOMENCLATURE PROJECT** 

3600: Research, Development, Test & Evaluation, Air Force PE 0602203F: Aerospace Propulsion 623145: Aerospace Power Technology

BA 2: Applied Research

B. Accomplishments/Planned Programs (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
<b>Description:</b> Develop technologies for special purpose applications, including hybrid electrical power, thermal management systems, and energy conversion/storage components and subsystems.					
FY 2011 Accomplishments:  Developed increased fuel flexibility and integrated energy harvesting technologies for expanded special purpose applications for improved power and energy density. Performed integrated flight-weight subsystems flight tests to demonstrate power and energy density goals.					
FY 2012 Plans:  Develop fully ruggedized hybrid power subsystems and energy harvesting components. Perform flight tests of these subsystems to demonstrate achievement of power and energy density goals for special purpose applications. Explore technology set for development of power systems for micro air vehicles. Note: In FY 2012, efforts in this thrust are increased due to higher AF priorities.					
FY 2013 Base Plans: Develop and test small and micro remote piloted aircraft power systems to provide enhanced capability and endurance and logistical fuel compatibility.					
FY 2013 OCO Plans: N/A					
Accomplishments/Planned Programs Subtotals	31.346	32.639	30.061	-	30.061

## C. Other Program Funding Summary (\$ in Millions)

		<del></del>	FY 2013	FY 2013	FY 2013					Cost To	
<u>Line Item</u>	FY 2011	FY 2012	Base	OCO	<u>Total</u>	FY 2014	FY 2015	FY 2016	FY 2017	Complete	<b>Total Cost</b>
• N/A: <i>N/A</i>	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	Continuing	Continuing

## D. Acquisition Strategy

N/A

## E. Performance Metrics

Please refer to the Performance Base Budget Overview Book for information on how Air Force resources are applied and how those resources are contributing to Air Force performance goals and most importantly, how they contribute to our mission.

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force										ruary 2012		
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research								PROJECT 624847: Rocket Propulsion Technology				
COST (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	FY 2014	FY 2015	FY 2016	Cost To Complete	Total Cost		
624847: Rocket Propulsion Technology	56.966	60.390	55.293	-	55.293	54.888	54.689	54.727	59.374	Continuing	Continuing	

## A. Mission Description and Budget Item Justification

B. Accomplishments/Planned Programs (\$ in Millions)

This project develops rocket propulsion technologies for space access, space maneuver, missiles, the sustainment of strategic systems (including solid boost/ missile propulsion, post boost control, aging and surveillance efforts), and tactical missiles. Analytical and experimental areas of emphasis are propellants, propellant management, combustion, rocket material applications, technology for sustainment of strategic systems, and innovative space propulsion concepts. Technologies of interest will improve reliability, performance, survivability, affordability, and environmental compatibility of these systems. Technologies are developed to reduce the weight and cost of components using new materials and improved designs and manufacturing techniques. All efforts in this project contribute to the sustainment of the rocket propulsion industry, providing rocket propulsion technology for the entire Department of Defense. Technologies developed under this program enable capabilities of interest to both the Department of Defense and NASA. Efforts include modeling and simulation, proof of concept tests of critical components, advanced component development, and ground-based tests. Aging and surveillance efforts could reduce lifetime prediction uncertainties for individual motors by 50 percent, enabling motor replacement for cause.

D. Accomplishments/r famed r rograms (\$\psi\$ in \text{winnons})	FY 2011	FY 2012	Base	OCO	Total
Title: Major Thrust 1.	3.838		6.208		6.208
<b>Description:</b> Develop, characterize, and test advanced hydrocarbons, energetics, solid propellants, and monopropellants to increase space launch payload capability and refine new synthesis methods.					
FY 2011 Accomplishments:  Conducted experimental and analytical evaluation of potential hydrocarbon fuel additives to improve performance of kerosene. Continued synthesis and downselect process and scale-up of promising high energy-density materials candidates. Evaluated scaled-up propellants in advanced combustion devices to determine materials compatibility and performance to include supporting large-scale motor tests. Explored and developed advanced ionic liquids. Continued scale-up experiments of promising ionic liquids for further characterization. Continued proof of concept for new computational code to predict molecular properties of various promising propellant ingredients. Continued evaluation of suitability for ionic liquid propellants for missile defense interceptor and spacecraft propulsion demonstrations. Continued technology transfer to industry for production of downselected propellants. Continued high performance bi-propellant identification and synthesis program.  FY 2012 Plans:					

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FY 2013 | FY 2013 | FY 2013

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			D	ATE: Febru	ary 2012	
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research	R-1 ITEM NOMENCLATURE PE 0602203F: Aerospace Propulsion	PROJECT 624847: Rocket Propulsion Technology				
B. Accomplishments/Planned Programs (\$ in Millions)		FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
Conduct experimental and analytical evaluation of potential hydrocal of kerosene. Continue synthesis and downselect process and scale materials candidates. Evaluate scaled-up propellants in advanced compatibility and performance to include supporting large-scale molionic liquids including synthesis and characterization. Continue scale for further characterization. Continue evaluation of suitability for ion interceptor and spacecraft propulsion demonstrations. Continue technologies and continue technologies are continued to the continue technologies.	t-up of promising high energy-density combustion devices to determine materials for tests. Explore and develop advanced e-up experiments of promising ionic liquids ic liquid propellants for missile defense hnology transfer to industry for production of					
FY 2013 Base Plans: Conduct experimental and analytical evaluation of potential hydroca of kerosene. Continue synthesis and downselect process and scale materials candidates. Evaluate scaled-up propellants in advanced compatibility and performance to include supporting large-scale mo ionic liquids including synthesis and characterization. Continue scal for further characterization. Continue evaluation of suitability for ion interceptor and spacecraft propulsion demonstrations. Continue tec downselected propellants. Continue high performance bi-propellants						
FY 2013 OCO Plans: N/A						
Title: Major Thrust 2.		7.988	7.364	7.766	-	7.766
<b>Description:</b> Develop advanced liquid engine combustion technolo preserving chamber lifetime and reliability needs for engine uses in						
FY 2011 Accomplishments: Characterized, studied, and evaluated shear injector performance to and prevent damage to engines. Validated study results in more reat transition of predictive tools to industry. Developed, analyzed, and to technology, including injectors and chambers. Developed improved and fluid flow/heat transfer processes leading to new methodologies combustion instabilities in hydrocarbon fueled liquid rocket engines numbers of costly full-scale component and engine tests. Character	alistic rocket-chamber conditions and begin ransitioned advanced combustion device understanding of fundamental combustion is for thermal management, scaling, and reducing the need for conducting large					

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force

APPROPRIATION/BUDGET ACTIVITY

3600: Research, Development, Test & Evaluation, Air Force
BA 2: Applied Research

BA 2: Applied Research

DATE: February 2012

R-1 ITEM NOMENCLATURE
PE 0602203F: Aerospace Propulsion
624847: Rocket Propulsion Technology

### B. Accomplishments/Planned Programs (\$ in Millions) FY 2013 FY 2013 FY 2013 FY 2011 FY 2012 Base OCO Total rig in preparation for evaluating cooling channel designs. Conducted validation and verification of advanced modeling and simulation capabilities. Performed pre-selection of most promising advanced propulsion concepts; applied realistic computational models to optimize performance. Refined experimental demonstrations of proofof-concepts; continued development of realistic computational models. Conducted system trade studies with improved performance models to evaluate potential return on investment. FY 2012 Plans: Using data obtained from a hydrocarbon demonstrator engine, characterize, study, and evaluate injector performance to ensure chamber/injector compatibility and prevent damage to engines. Validate study results in more realistic rocket-chamber conditions and transition of predictive tools to industry. Feed advanced combustion device technology into hydrocarbon boost efforts, continue additional analysis on changing designs and concepts. Develop improved understanding of fundamental combustion and fluid flow/heat transfer processes leading to new methodologies for thermal management, scaling, and combustion instabilities in hydrocarbon fueled liquid rocket engines, reducing the need for conducting large numbers of costly full-scale component and engine tests. Evaluate novel nozzle cooling channels for use with hydrocarbon fuels in the high heat flux test rig. Conduct validation and verification of advanced modeling and simulation capabilities. Perform pre-selection of most promising advanced propulsion concepts; apply realistic computational models to optimize performance. Refine experimental demonstrations of proof-of-concepts, continue development of realistic computational models. Conduct system trade studies with improved performance models to evaluate potential return on investment. FY 2013 Base Plans: Using data obtained from a hydrocarbon demonstrator engine, characterize, study, and evaluate injector performance to ensure chamber/injector compatibility and prevent damage to engines. Validate study results in more realistic rocket-chamber conditions and transition of predictive tools to industry. Begin efforts looking at multi-injector designs and control effectors. Feed advanced combustion device technology into a hydrocarbon boost demo and to various contractor designs, continue additional analysis on changing designs and concepts. Develop improved understanding of fundamental combustion and fluid flow/heat transfer processes leading to new methodologies for thermal management, scaling, and combustion instabilities in hydrocarbon fueled liquid

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rocket engines, reducing the need for conducting large numbers of costly full-scale component and engine tests. Evaluate novel nozzle cooling channels for use with hydrocarbon fuels in the high heat flux test rig. Conduct validation and verification of advanced modeling and simulation capabilities. Perform pre-selection of most promising advanced propulsion concepts; apply realistic computational models to optimize performance. Refine experimental demonstrations of proof-of-concepts, continue development of realistic computational models.

Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force  IPPROPRIATION/BUDGET ACTIVITY  600: Research, Development, Test & Evaluation, Air Force IA 2: Applied Research  IS. Accomplishments/Planned Programs (\$ in Millions)  Conduct system trade studies with improved performance models to evaluate potential return on investment.  Begin development of modeling tools for characterization of rocket plumes in air and space.  IVA  Title: Major Thrust 3.  Description: Develop advanced material applications for lightweight components and material property enhancements for current and future rocket propulsion systems.  IVY 2011 Accomplishments:  Developed new advanced ablative components using hybrid polymers. Continued to characterize and finalize processing parameters of new nano-reinforced high temperature polymers and scale-up processing of carbonarbon materials. Developed new advanced materials for use with high-energy propellants. Continued to explore applications of nanocomposites for the hydrocarbon boost demo and other liquid rocket engine components and ptimize processing technology using multifunctional nanomaterials. Continued to characterize and understand the mechanisms behind a new class of hydrophobic and oleophobic materials exploring various transition apportunities.		ROJECT 4847: <i>Rock</i> FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
Research, Development, Test & Evaluation, Air Force A 2: Applied Research  Conduct system trade studies with improved performance models to evaluate potential return on investment. Begin development of modeling tools for characterization of rocket plumes in air and space.  FY 2013 OCO Plans:  A/A  Citle: Major Thrust 3.  Description: Develop advanced material applications for lightweight components and material property enhancements for current and future rocket propulsion systems.  FY 2011 Accomplishments:  Developed new advanced ablative components using hybrid polymers. Continued to characterize and finalize processing parameters of new nano-reinforced high temperature polymers and scale-up processing of carbonarbon materials. Developed new advanced materials for use with high-energy propellants. Continued to explore applications of nanocomposites for the hydrocarbon boost demo and other liquid rocket engine components and primize processing technology using multifunctional nanomaterials. Continued to characterize and understand the mechanisms behind a new class of hydrophobic and oleophobic materials exploring various transition	62 <sup>4</sup> FY 2011	4847: Rock FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
Conduct system trade studies with improved performance models to evaluate potential return on investment. Begin development of modeling tools for characterization of rocket plumes in air and space.  FY 2013 OCO Plans:  W/A  Fitle: Major Thrust 3.  Description: Develop advanced material applications for lightweight components and material property enhancements for current and future rocket propulsion systems.  FY 2011 Accomplishments:  Developed new advanced ablative components using hybrid polymers. Continued to characterize and finalize processing parameters of new nano-reinforced high temperature polymers and scale-up processing of carbonarbon materials. Developed new advanced materials for use with high-energy propellants. Continued to explore applications of nanocomposites for the hydrocarbon boost demo and other liquid rocket engine components and aptimize processing technology using multifunctional nanomaterials. Continued to characterize and understand the mechanisms behind a new class of hydrophobic and oleophobic materials exploring various transition			Base	oco	Total
Regin development of modeling tools for characterization of rocket plumes in air and space.  FY 2013 OCO Plans:  WA  Fitle: Major Thrust 3.  Pescription: Develop advanced material applications for lightweight components and material property enhancements for current and future rocket propulsion systems.  FY 2011 Accomplishments:  Developed new advanced ablative components using hybrid polymers. Continued to characterize and finalize processing parameters of new nano-reinforced high temperature polymers and scale-up processing of carbon-carbon materials. Developed new advanced materials for use with high-energy propellants. Continued to explore applications of nanocomposites for the hydrocarbon boost demo and other liquid rocket engine components and aptimize processing technology using multifunctional nanomaterials. Continued to characterize and understand the mechanisms behind a new class of hydrophobic and oleophobic materials exploring various transition	5.492	5.722	1.000	) -	1.000
Fitle: Major Thrust 3.  Description: Develop advanced material applications for lightweight components and material property enhancements for current and future rocket propulsion systems.  FY 2011 Accomplishments: Developed new advanced ablative components using hybrid polymers. Continued to characterize and finalize processing parameters of new nano-reinforced high temperature polymers and scale-up processing of carbon-materials. Developed new advanced materials for use with high-energy propellants. Continued to explore applications of nanocomposites for the hydrocarbon boost demo and other liquid rocket engine components and applications of processing technology using multifunctional nanomaterials. Continued to characterize and understand the mechanisms behind a new class of hydrophobic and oleophobic materials exploring various transition	5.492	5.722	1.000	) -	1.000
Description: Develop advanced material applications for lightweight components and material property enhancements for current and future rocket propulsion systems.  FY 2011 Accomplishments: Developed new advanced ablative components using hybrid polymers. Continued to characterize and finalize processing parameters of new nano-reinforced high temperature polymers and scale-up processing of carbonarbon materials. Developed new advanced materials for use with high-energy propellants. Continued to explore applications of nanocomposites for the hydrocarbon boost demo and other liquid rocket engine components and aptimize processing technology using multifunctional nanomaterials. Continued to characterize and understand the mechanisms behind a new class of hydrophobic and oleophobic materials exploring various transition	5.492	5.722	1.000	-	1.000
Explain the complishments:  Developed new advanced ablative components using hybrid polymers. Continued to characterize and finalize processing parameters of new nano-reinforced high temperature polymers and scale-up processing of carbonarbon materials. Developed new advanced materials for use with high-energy propellants. Continued to explore applications of nanocomposites for the hydrocarbon boost demo and other liquid rocket engine components and aptimize processing technology using multifunctional nanomaterials. Continued to characterize and understand the mechanisms behind a new class of hydrophobic and oleophobic materials exploring various transition					
Developed new advanced ablative components using hybrid polymers. Continued to characterize and finalize processing parameters of new nano-reinforced high temperature polymers and scale-up processing of carbonarbon materials. Developed new advanced materials for use with high-energy propellants. Continued to explore applications of nanocomposites for the hydrocarbon boost demo and other liquid rocket engine components and optimize processing technology using multifunctional nanomaterials. Continued to characterize and understand the mechanisms behind a new class of hydrophobic and oleophobic materials exploring various transition					
pportunities.					
Pevelop new material formulations that better address the challenges inside solid rockets. Continue to characterize and finalize processing parameters of new reinforced high temperature polymers and scale-up processing of carbon-carbon materials. Refine formulations of polymers for use in various liquid rocket engine components. Continue to characterize and understand the mechanisms behind a new class of hydrophobic and eleophobic materials exploring various transition opportunities.					
Pevelop new material formulations that better address the challenges inside solid rockets. Continue to characterize and finalize processing parameters of new reinforced high temperature polymers and scale-up processing of carboncarbon materials. Refine formulations of polymers for use in various liquid rocket engine components. Continue to characterize and understand the mechanisms behind a new class of hydrophobic and eleophobic materials exploring various transition opportunities. Decrease in FY 2013 due to higher Department of Defense priorities.					

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			D	ATE: Febru	ary 2012		
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research	R-1 ITEM NOMENCLATURE PE 0602203F: Aerospace Propulsion	PROJECT 624847: Rocket Propulsion Technology					
B. Accomplishments/Planned Programs (\$ in Millions)		FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	
N/A							
Title: Major Thrust 4.		24.437	17.103	11.345	-	11.345	
<b>Description:</b> Develop advanced liquid engine technologies for impreliability needs for engine uses in expendable and reusable launc							
FY 2011 Accomplishments:  Developed enabling hydrocarbon boost technology for future space activities. Continued development of engine health monitoring tech technology development effort. Developed advanced hydrocarbon kerosene. Developed and demonstrated in-house, moderate scale Developed high performance compact liquid rocket engine technologie bipropellant technologies.	engine technologies using fuels other than liquid rocket component testing capability.						
FY 2012 Plans: Continue development of enabling hydrocarbon boost technology of reduction activities for the development of hydrocarbon boost technologies technologies supporting the hydrocarbon boost to advanced hydrocarbon engine technologies using fuels other than house, moderate scale liquid rocket component testing capability. Trocket engine technologies. Continue development and evaluation 2012, efforts in this thrust are decreased due to higher AF priorities.							
FY 2013 Base Plans: Develop enabling hydrocarbon boost technology for future spacelif activities for the development of hydrocarbon boost technologies. Of monitoring technologies supporting the hydrocarbon boost technologies using fuels other than kerosene. Scale liquid rocket component testing capability. Develop high perfetechnologies. Continue development and evaluation of bipropellant thrust are decreased due to higher AF priorities.	Continue development of engine health ogy development effort. Develop advanced Develop and demonstrate in-house, moderate ormance compact liquid rocket engine						
FY 2013 OCO Plans:							
- 1 2010 000 1 <b>18110</b> 1		I	I		I	I	

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			DATE: February 2012
APPROPRIATION/BUDGET ACTIVITY	R-1 ITEM NOMENCLATURE	PROJECT	
3600: Research, Development, Test & Evaluation, Air Force	PE 0602203F: Aerospace Propulsion	624847: Ro	ocket Propulsion Technology
BA 2: Applied Research			

B. Accomplishments/Planned Programs (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
N/A					
Title: Major Thrust 5.	5.391	4.930	7.580	-	7.580
<b>Description:</b> Develop solar electric, solar thermal, chemical, and advanced propulsion technologies for station-keeping, repositioning, and orbit transfer for satellites and satellite constellations.					
FY 2011 Accomplishments:  Evaluated advanced plasma thrusters for microsatellites propulsion systems. Scale-up tested monopropellants, evaluated advanced ignition schemes and chamber concepts. Assessed advanced chemical propulsion technology developments for satellite thrusters; continued component developments. Developed advanced multi-mode chemical-electric propulsion concepts for satellites; continued component developments. Developed next generation high power electric spacecraft propulsion. Continued advanced modeling and simulation tool developments to improve design and analysis tools for a wide range of spacecraft propulsion concepts/ technologies.					
FY 2012 Plans: Characterize advanced plasma thrusters for microsatellites propulsion systems. Conduct scale-up of advanced monopropellants, evaluate advanced ignition schemes and chamber concepts. Assess advanced chemical propulsion technology developments for satellite thrusters and continue component developments. Develop advanced multi-mode chemical-electric propulsion concepts for satellites and continue component developments. Continue development of next generation high power electric spacecraft propulsion. Continue advanced modeling and simulation tool developments to improve design and analysis tools for a wide range of spacecraft propulsion concepts/technologies.					
FY 2013 Base Plans: Continue characterization of advanced plasma thrusters for microsatellites propulsion systems. Conduct scale-up of advanced monopropellants, evaluate advanced ignition schemes and chamber concepts. Assess advanced chemical propulsion technology developments for satellite thrusters and continue component developments. Develop advanced multi-mode chemical-electric propulsion concepts for satellites and continue component developments. Continue development of next generation high power electric spacecraft propulsion. Continue advanced modeling and simulation tool developments to improve design and analysis tools for a wide range of spacecraft propulsion concepts/technologies. Begin exploration into new generation of bipropellant					

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			D	ATE: Febru	ary 2012	
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research	R-1 ITEM NOMENCLATURE PE 0602203F: Aerospace Propulsion		ROJECT 24847: Rock	et Propulsic	n Technolo	gy
B. Accomplishments/Planned Programs (\$ in Millions)		FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
spacecraft thruster technologies. Note: In FY 2013, efforts in this thrupriorities.	ust are increased due to higher AF					
FY 2013 OCO Plans: N/A						
Title: Major Thrust 6.		7.791	14.884	11.784	-	11.784
<b>Description:</b> Develop missile propulsion and boost technologies for	space access and strike applications.					
FY 2011 Accomplishments:  Continued the component development and risk reduction efforts for Demonstrated components for solid rocket motors. Developed advant Continued development and evaluation of next generation of updated analysis tools for missile propulsion components and applications. Conforts.						
FY 2012 Plans: Test components as part of risk reduction efforts for future missile protactical propulsion technologies. Continue development and evaluation based modeling, simulation, and analysis tools for missile propulsion	on of next generation of updated, physics-					
FY 2013 Base Plans: Develop advanced tactical propulsion technologies. Continue develo of updated, physics-based modeling, simulation, and analysis tools for applications.						
FY 2013 OCO Plans: N/A						
Title: Major Thrust 7.		2.029	5.444	9.610	-	9.610
<b>Description:</b> Develop missile propulsion technologies and aging and missiles.	I surveillance technologies for ballistic					
FY 2011 Accomplishments: Conducted advanced service life prediction technology program. Dev sensors to be attached to solid rocket motors and tools that can integ						

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**DATE:** February 2012 Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force **R-1 ITEM NOMENCLATURE PROJECT** APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force PE 0602203F: Aerospace Propulsion 624847: Rocket Propulsion Technology BA 2: Applied Research B. Accomplishments/Planned Programs (\$ in Millions) FY 2013 FY 2013 FY 2013 FY 2011 FY 2012 **Base** OCO Total and surveillance tool suite. Continued efforts to integrate advanced aging and surveillance technologies into demonstrations to validate and verify efforts to reduce uncertainties and accurately model motor behavior. Continued development of next generation of chemical and aging mechanism modeling, simulation, and analysis tools, sensor schemes and tools, and non-destructive analysis tools. FY 2012 Plans: Conduct sub-scale testing of existing and advanced sensors to be attached to solid rocket motors and tools that can integrate sensor data into existing aging and surveillance tool suite. Integrate advanced aging and surveillance technologies into demonstrations to validate and verify efforts to reduce uncertainties and accurately model motor behavior. Apply next generation of chemical and aging mechanism modeling, simulation, and analysis tools, sensor schemes and tools, and non-destructive analysis tools. FY 2013 Base Plans: Conduct sub-scale testing of existing and advanced sensors to be attached to solid rocket motors and tools that can integrate sensor data into existing aging and surveillance tool suite. Integrate advanced aging and surveillance technologies into demonstrations to validate and verify efforts to reduce uncertainties and accurately

#### FY 2013 OCO Plans:

N/A

Accomplishments/Planned Programs Subtotals

56.966 60.390 55.293 -

55.293

## C. Other Program Funding Summary (\$ in Millions)

thrust are increased due to higher AF priorities.

model motor behavior. Apply next generation of chemical and aging mechanism modeling, simulation, and analysis tools, sensor schemes and tools, and non-destructive analysis tools. Note: In FY 2013, efforts in this

FY 2013 Cost To FY 2013 FY 2013 Line Item FY 2011 FY 2012 **Base** OCO FY 2014 FY 2015 **FY 2016** FY 2017 Complete Total Cost Total • N/A: N/A 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 Continuing Continuing

## D. Acquisition Strategy

N/A

PE 0602203F: Aerospace Propulsion

Air Force

UNCLASSIFIED

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force		DATE: February 2012		
APPROPRIATION/BUDGET ACTIVITY				
3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research	PE 0602203F: Aerospace Propulsion	PROJECT 624847: Rocket Propulsion Technology		
E. Performance Metrics				
Please refer to the Performance Base Budget Overview Book for Force performance goals and most importantly, how they contrib		ed and how those resources are contributing to		

PE 0602203F: Aerospace Propulsion Air Force

Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force  DATE: February 2012											
APPROPRIATION/BUDGET ACTI 3600: Research, Development, Tes BA 2: Applied Research	evelopment, Test & Evaluation, Air Force PE 0602203F: Aerospace Propulsion 625330: Aerospace Fuel Tec					el Technology	Y				
DA 2. Applied Research	I		I					I			
COST (\$ in Millions)	FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total	FY 2014	FY 2015	FY 2016	FY 2017	Cost To Complete	Total Cost
625330: Aerospace Fuel Technology	6.480	6.239	5.494	-	5.494	5.475	5.275	5.312	5.426	Continuing	Continuing

### A. Mission Description and Budget Item Justification

This project evaluates hydrocarbon-based fuels for legacy and advanced turbine engines, scramjets, pulse detonation and combined cycle engines. This project also considers fuel related concepts that can increase turbine engine operational reliability, durability, mission flexibility, energy efficiency, and performance while reducing weight, fuel consumption, and cost of ownership. Applications include missiles, aircraft, sustained high-speed vehicles, and responsive space launch. Analytical and experimental areas of emphasis include evaluations of fuel properties and characteristics of alternative fuels developed from unconventional sources (such as coal, natural gas, biomass, and combinations thereof), unique/alternate fuels and components used in integrated thermal and energy management systems including high heat sink fuel capability, fuels logistics and associated vulnerabilities, and combustion diagnostics and engine emissions measurements.

B. Accomplishments/Planned Programs (\$ in Millions)			FY 2013	FY 2013	FY 2013
	FY 2011	FY 2012	Base	oco	Total
Title: Major Thrust 1.	3.087	3.151	_	-	-
<b>Description:</b> Conduct evaluations and perform technical assessments of alternative hydrocarbon fuels derived from coal, natural gas, and biomass for use in legacy and advanced aerospace systems.					
FY 2011 Accomplishments:  Completed component "fit-for-purpose" evaluations of up to 100 percent synthetic paraffinic kerosene (SPK) and made recommendation as to maximum SPK in blend use. Completed initial evaluations of biomass derived aviation fuels and assessment of associated carbon dioxide footprint. Conducted follow-on component evaluations as available fuel quantities permit.					
FY 2012 Plans: Develop link between fully-synthetic fuel composition and basic physical properties and rig test performance.					
FY 2013 Base Plans: Decrease in FY 2013 due to higher Department of Defense priorities.					
FY 2013 OCO Plans: N/A					
Title: Major Thrust 2.	1.100	1.088	-	_	-

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Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force			D	ATE: Febru	ary 2012	
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research	R-1 ITEM NOMENCLATURE PE 0602203F: Aerospace Propulsion					
B. Accomplishments/Planned Programs (\$ in Millions)		FY 2011	FY 2012	FY 2013 Base	FY 2013 OCO	FY 2013 Total
<b>Description:</b> Develop and demonstrate advanced components and advanced aircraft integrated thermal and energy management syst	·					
FY 2011 Accomplishments: Assessed advanced aircraft thermal management designs. Develo thermal characteristics of aviation fuels used in integrated thermal advanced hydrocarbon based endothermic fuel technologies applic	and energy management systems. Developed					
FY 2012 Plans: Assess advanced catalyst approaches to enhancing heat sink in hy						
FY 2013 Base Plans: Decrease in FY 2013 due to higher Department of Defense priorities						
FY 2013 OCO Plans: N/A						
Title: Major Thrust 3.	Title: Major Thrust 3.				-	5.494
<b>Description:</b> Study and evaluate low-cost approaches to reduce for logistics vulnerabilities and develop detection and mitigation technology.						
FY 2011 Accomplishments: Assessed aberrant logistical fuels to support field operations and in Evaluated low-cost fuel additives and assessment of the impact on investigation of actions to mitigate the growth of biological agents in biological mutations in fuel leading to the development of resistance.	biological growth in fuel. Continued the n fuel. Investigated the development of					
FY 2012 Plans:  Develop biological growth mitigation approaches for commercial jecommercial off-the-shelf jet fuels. Evaluate approaches for portable field power generation.						
FY 2013 Base Plans:						

PE 0602203F: *Aerospace Propulsion* Air Force

Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force  DATE: February 20								
APPROPRIATION/BUDGET ACTIVITY 3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research	<b>R-1 ITEM NO</b> PE 0602203			<b>F</b>	space Fuel	Technology		
B. Accomplishments/Planned Programs (\$ in Millions)	•							
Assess impact of conversion to commercial jet fuel (without JP-8 additi systems. Note: In FY 2013, efforts in this thrust are increased due to he								
FY 2013 OCO Plans: N/A								
Title: Major Thrust 4.				1.29	3 1.000	-	-	-
<b>Description:</b> Develop and test advanced emissions diagnostic techniq Conduct evaluations of the combustion and emissions characteristics of			ion systems.					
FY 2011 Accomplishments:  Developed diagnostic protocols for aircraft ground emissions measurer evaluations on fielded engines to investigate particulate formation and diagnostics applicable to advanced high pressure combustor systems. biomass derived aviation fuels. Conducted assessment of combustion derived aviation fuels.								
FY 2012 Plans: Implement advanced particulate diagnostics in high-pressure combustor synthetic fuels relative to JP-8 and JP-8/synthetic blends.	s from fully-							
FY 2013 Base Plans: Decrease in FY 2013 due to higher Department of Defense priorities.								
FY 2013 OCO Plans: N/A								
Accomp	lishments/Plar	nned Progra	ıms Subtotals	6.48	0 6.239	5.494	-	5.494
C. Other Program Funding Summary (\$ in Millions)  FY 201	3 FY 2013	FY 2013					Cost To	
Line Item FY 2011 FY 2012 Bas		Total	FY 2014	FY 2015	FY 2016	FY 2017		Total Cost
• N/A: <i>N/A</i> 0.000 0.000 0.000	0.000	0.000	0.000	0.000	0.000	0.000	Continuing	Continuing
D. Acquisition Strategy N/A								

PE 0602203F: *Aerospace Propulsion* Air Force

Exhibit R-2A, RDT&E Project Justification: PB 2013 Air Force		<b>DATE</b> : February 2012
APPROPRIATION/BUDGET ACTIVITY	R-1 ITEM NOMENCLATURE	PROJECT
3600: Research, Development, Test & Evaluation, Air Force BA 2: Applied Research	PE 0602203F: Aerospace Propulsion	625330: Aerospace Fuel Technology
E. Performance Metrics  Please refer to the Performance Base Budget Overview Book for information on how Air Force resources are applied and how those resources are contributing to Air Force performance goals and most importantly, how they contribute to our mission.		

PE 0602203F: Aerospace Propulsion Air Force